TECHNICAL REPORT IGE-293

A USER GUIDE FOR TRIVAC VERSION4

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SUMMARY

TRIVAC is a computer code intended to compute the neutron flux in a fractional or in a full core representation of a nuclear reactor. Interested readers can obtain fundamental informations about fullcore calculations in Chapter 5 of Ref. 1. The multigroup and multidimensional form of the diffusion equation or simplified P_n equation is first discretized to produce a consistent matrix system. This matrix system is subsequently solved using iterative techniques (inverse or preconditioned power method with ADI preconditioning) and sparse matrix algebra techniques (triangular factorization). The actual implementation of TRIVAC allows the discretization of 1-D geometries (slab and cylindrical), 2-D geometries (Cartesian, cylindrical and hexagonal) and 3-D geometries (Cartesian and hexagonal). Many discretization techniques are available, including mesh corner or mesh centered finite difference methods, collocation techniques of various order and finite element methods based on a primal or dual functional formulation. TRIVAC also permits the equations of the generalized perturbation theory (GPT) to be solved as fixed source eigenvalue problems. Finally, several implicit numerical schemes are available for the solving of space-time neutron kinetics problems.

The execution of TRIVAC is controlled by the generalized GAN driver.^[2] It is modular and can be interfaced easily with other production codes.

Contents

Copyright Notice for TRIVAC				
Contents	Contents			
List of Fig	List of Figures			
List of Tal	ples			
1 INPU	IT DATA SPECIFICATIONS			
1.1	Syntactic rules for input data specifications			
1.2	The global input structure			
1.3	The GEO: module			
1.0	1.3.1 Data input for module GEO:			
	1.3.2 Examples of geometries			
1.4	The MAC: module			
	1.4.1 Data input for module MAC:			
	142 Description of the nuclear data 15			
1.5	The BIVACT: module 18			
1.0	1.5.1 Data input for module BIVACT:			
1.6	The TRIVAT: module 21 module 21 module 21			
1.0	1.6.1 Data input for module TRIVAT:			
17	The BIVACA: module 24			
1.1	171 Deta input for module BIVACA:			
1.8	The TRIVAA: module 25			
1.0	1.8.1 Data input for module TPIVA.			
1.0	The EUD: module			
1.9	1.0.1 Deta input for module ELID:			
1 10	The DELTA: module 200			
1.10	1 10 1 Deta input for modulo DELTA:			
1 1 1	The CDTELLI, module 2017			
1.11	1 11 1 Data input for module CDTELU.			
1 1 9	The OUT: module 324			
1.12	110 Uo1: Inodule			
1 1 9	1.12.1 Data input for module 001:			
1.13	The INTERNA medicle 20			
1.14	Ine INIKIN: module 38 1141 D 1 1 1 20			
1.15	1.14.1 Data input for module INIKIN:			
1.15	1151 D			
1.10	1.15.1 Data input for module KINSUL:			
1.10	The VAL: module 44 1161 D D 44			
	1.10.1 Data input for module VAL:			
2 EXA	MPLES OF INPUT DATA FILES			
2.1	IAEA-2D benchmark			
2.2	Biblis-2D benchmark			
2.3	IAEA-3D benchmark			
2.4	S30 hexagonal benchmark in 2-D			
2.5	LMW benchmark in 2-D			
References				
Index				

List of Figures

1	The TRIVAC modular approach
2	Hexagonal geometries of type S30 and SA60
3	Hexagonal geometries of type SB60 and S90
4	Hexagonal geometries of type R120 and R180
5	Hexagonal geometry of type SA180
6	Hexagonal geometry of type SB180
7	Hexagonal geometry of type COMPLETE 8
8	Cylindrical correction in Cartesian geometry
9	Slab geometry with mesh-splitting 12
10	Two-dimensional hexagonal geometry 12
11	Description of the IAEA-2D benchmark
12	Description of the Biblis-2D benchmark, rods-withdrawn configuration
13	Description of the IAEA-3D benchmark
14	Description of the S30 hexagonal benchmark
15	Description of the LMW benchmark in 2-D

List of Tables

1	Structure (TRIVAC)	1
2	Structure (GEO:)	2
3	Structure (geo_data1)	3
4	Structure (geo_data2)	3
5	Structure (descBC)	4
6	Structure (descMC)	9
7	Structure (descPOS)	0
8	Structure (MAC:)	3
9	Structure (mac_data)	3
10	Structure (macxs)	5
11	Structure (BIVACT:)	8
12	Structure (bivact_data) 1	8
13	Structure (TRIVAT:)	1
14	Structure (trivat_data) 2	1
15	Structure (BIVACA:)	4
16	Structure (bivaca_data) 2	4
17	Structure (TRIVAA:)	5
18	Structure (trivaa_data) 2	5
19	Structure (FLUD:)	7
20	Structure (flud_data)	7
21	Structure (DELTA:)	0
22	Structure (delta_data)	0
23	Structure (GPTFLU:)	2
24	Structure (gptflu_data)	2
25	Structure (OUT:)	4
26	Structure (out_data)	4
27	Structure (ERROR:)	6
28	Structure (INIKIN:)	8
29	Structure (inikin_data)	8
30	Structure (KINSOL:)	:1
31	Structure (kinsol_data)	:2
32	Structure (VAL:)	4
33	Structure (descval)	:4

1 INPUT DATA SPECIFICATIONS

1.1 Syntactic rules for input data specifications

The input data to any module is read in free format using the subroutine **REDGET**. The rules for specifying the input data are therefore given in this section. The users guide was written using the following conventions:

- the parameters surrounded by single square brackets '[]' denote an optional input;
- the parameters surrounded by double square brackets '[[]]' denote an optional input which may be repeated as many times as desired;
- the parameters in braces separated by vertical bars '{ | | }' denote a choice of input where (one and *only* one is mandatory);
- the parameters in **bold face** and in brackets '()' denote an input structure;
- the parameters in italics and in brackets with an index '(*data*(i), i=1,n)' denote a set of n inputs;
- the words using the typewriter font are character constants keywordS used as keywords;
- the words in italics are user defined variables, they should be lower case and are of type integer (starting with i to n) and real (starting with a to h or o to z) or of type character in uppercase CHARACTER.

1.2 The global input structure

TRIVAC is built around the GAN generalized driver.^[2] Input data must therefore follow the calling specifications given below:

Table 1: Structure (TRIVAC)

```
[ LINKED_LIST [[ NAME1 ]] ; ]
[ XSM_FILE [[ NAME2 ]] ; ]
[ SEQ_BINARY [[ NAME3 ]] ; ]
[ SEQ_ASCII [[ NAME4 ]] ; ]
[ MODULE [[ NAME5 ]] ; ]
[[ (specif) ]]
END: ;
```

where

- NAME1 Character*12 name of a LCM object.
- NAME2 Character*12 name of an XSM file.
- *NAME3* Character*12 name of a sequential binary file.
- *NAME4* Character*12 name of a sequential ASCII file.
- NAME5 Character*12 name of a module.
- (specif) Input specifications for a single module. Specifications for TRIVAC modules will be given in the following sections.

The input data always begin with the declaration of each LCM object, XSM file, sequential (binary or ASCII) file that will be required by the following modules. This is followed by the declaration of the modules actually used in the input data deck. The following data describe a sequence of module calls, in the format of the GAN generalized driver. As indicated in Fig. 1, the modules communicate with each other throught LCM objects or XSM files whose specifications are given in section 2. The TRIVAC user generally have the choice to declare its data structures as LINKED_LIST to reduce CPU time resources or as XSM_FILE to reduce CPU memory resources.

The input data always end with a call to the END: module.



Figure 1: The TRIVAC modular approach.

1.3 The GEO: module

The GEO: module is used to create or modify a geometry. The geometry definition module in TRIVAC permits all the characteristics (coordinates, material mixture type indices and boundary conditions) of a simple or complex geometry to be specified. The method used to specify the geometry is independent of the discretization module to be used subsequently. Each geometry is represented by a name (character*12) and is saved in a LCM object or an XSM file under its given name. It is always possible to modify a given existing geometry or copy it into a neighbouring LCM object under a new name. The calling specifications are:

Table 2: Structure (GEO:)

{ GEOM1 := GEO: :: (geo_data1) | GEOM1 := GEO: { GEOM1 | GEOM2 } :: (geo_data2) }

where

GEOM1 character*12 name of the LCM object (type L_GEOM) that will contain the geometry.

GEOM2 character*12 name of a LCM object (type L_GEOM) containing the existing geometry. The type and all the characteristics of GEOM2 will be copied onto GEOM1.

(geo_data1) structure describing the characteristics of a new geometry (see Sect. 1.3.1).

(geo_data2) structure describing the change to the characteristics of an existing geometry (see Sect. 1.3.1).

 $1.3.1 \ Data \ input \ for \ module \ \texttt{GE0}$:

Structures (geo_data1) and (geo_data2) serve to define the principle components of a geometry (dimensions, materials, boundary conditions):

Table 3: Structure (geo_data1)

```
{ HOMOGE | CAR1D lx | TUBE lr | SPHERE lr | CAR2D lx ly | TUBEZ lr lz | CAR3D lx ly lz |
    HEX lh | HEXZ lh lz }
[ EDIT iprint ]
(descBC)
(descMC)
(descPOS)
;
```

Table 4: Structure (geo_data2)

[EDIT iprint] (descBC) (descMC) (descPOS) ;

where

HOMOGE	infinite homogeneous geometry.
CAR1D	one dimensional plane geometry (infinite slabs).
TUBE	cylindrical geometry (infinite tubes or cylinders).
SPHERE	spherical geometry (concentric spheres).
CAR2D	two-dimensional cartesian geometry.
TUBEZ	polar geometry $(R-Z)$.
CAR3D	three-dimensional cartesian geometry.
HEX	two-dimensional hexagonal geometry.
HEXZ	three-dimensional hexagonal geometry.
lx	number of subdivisions along the X axis (before mesh-splitting).

ly	number of subdivisions along the Y axis (before mesh-splitting).
lz	number of subdivisions along the Z axis (before mesh-splitting).
lr	number of cylinders or spherical shells (before mesh-splitting).
lh	number of hexagons in an axial plane (including the virtual hexagons).
EDIT	keyword used to set <i>iprint</i> .
iprint	index used to control the printing in module $GE0:$. =0 for no print; =1 for minimum printing (default value); =2 for printing the geometry state vector.
(descBC)	structure allowing the boundary conditions surrounding the geometry to be treated.
(descMC)	structure allowing material mixtures to be associated with a geometry.
(descPOS)	structure allowing the coordinates of a geometry to be described.
(T) · ·	

The inputs corresponding to the (\mathbf{descBC}) structure are the following:

Table 5: Structure (descBC)

[X-{VOID REFL DIAG TRAN SYME ALBE{albedo icode}]ZERO
$ $ CYLI $ $ ACYL $\{ albedo icode \} \}$
[X+ { VOID REFL DIAG TRAN SYME ALBE { albedo icode } ZERO
CYLI ACYL { albedo icode } }]
[Y-{VOID REFL DIAG TRAN SYME ALBE{albedo icode}]ZERO
CYLI ACYL { albedo icode } }]
[Y+ { VOID REFL DIAG TRAN SYME ALBE { albedo icode } ZERO
CYLI ACYL { albedo icode } }]
[Z-{VOID REFL TRAN SYME ALBE{albedo icode}]ZERO}]
[Z+{VOID REFL TRAN SYME ALBE{albedo icode}]ZERO}]
$[R+ \{ VOID REFL ALBE \{ albedo icode \} ZERO \}]$
[HBC { S30 SA60 SB60 S90 R120 R180 SA180 SB180 COMPLETE }
$\{ VOID REFL SYME ALBE \{ albedo icode \} ZERO \}]$
[RADS [ANG] nrads (xrad(ir), rrad(ir) [, ang(ir)], ir=1, nrads)]

where

Х-	negative X side.
Y-	negative Y side.
Z-	negative Z side.
Х+	positive X side.
Y+	positive Y side.
Z+	positive Z side.
R+	side surrounding cylinders or spheres.
HBC	side surrounding a hexagonal geometry.
VOID	the side under consideration has a zero incoming current boundary condition.

- DIAG the side under consideration is external to a diagonal axis of symmetry.
- TRAN the side under consideration is connected to the opposite side of the domain. This option permits a translation condition to be treated.
- SYME the side under consideration is next to an axial axis of symmetry. (symmetric with respect to the central axis of the last row of volumes). The SYME condition can also be used in hexagonal geometry, but only with S30 and SA60 symmetries.
- ALBE the side under consideration has an arbitrary albedo to be specified.
- albedo geometrical albedo corresponding to the boundary condition ALBE (albedo ≥ 0.0).
- *icode* index of a physical albedo corresponding to the boundary condition ALBE. The numerical values of the physical albedo are supplied by the module MAC:.
- ZER0 the side under consideration has a zero flux boundary condition.
- CYLI the side under consideration has a zero incoming current boundary condition with a circular correction applied on the Cartesian boundary. This option is only available in the X-Y plane for CAR2D and CAR3D geometries defined for TRIVAC full-core calculations.
- ACYL the side under consideration has an arbitrary albedo with a circular correction applied on the Cartesian boundary. This option is only available in the X-Y plane for CAR2D and CAR3D geometries defined for TRIVAC full-core calculations.
- s30 hexagonal symmetry of one twelfth of an assembly (see Fig. 2).



Figure 2: Hexagonal geometries of type S30 and SA60

SA60 hexagonal	symmetry of	one sixth of	an assembly	of type A	(see Fig. 2)
----------------	-------------	--------------	-------------	-----------	---------------	---

SB60 hexagonal symmetry of one sixth of an assembly of type B (see Fig. 3).

- S90 hexagonal symmetry of one quarter of an assembly (see Fig. 3).
- R120 hexagonal symmetry of one third of an assembly (rotational symmetry) (see Fig. 4).
- R180 rotational symmetry of a half assembly (see Fig. 4).
- SA180 hexagonal symmetry of half a type A assembly (see Fig. 5).
- SB180 hexagonal symmetry of half a type B assembly (see Fig. 6).
- COMPLETE complete hexagonal assembly (see Fig. 7).



Figure 3: Hexagonal geometries of type SB60 and S90



Figure 4: Hexagonal geometries of type R120 and R180 $\,$



Figure 5: Hexagonal geometry of type SA180



Figure 6: Hexagonal geometry of type SB180



Figure 7: Hexagonal geometry of type COMPLETE

- **RADS** This keyword is used to specify the cylindrical correction applied in the X Y plane for CAR2D and CAR3D geometries.^[12]
- ANG This keyword allows the angle (see Fig. 8) of the cylindrical notch to be set. By default, no notch is present.
- *nrads* Number of different corrections along the cylinder main axis (i.e. the Z axis).
- xrad(ir) Coordinate of the Z axis from which the correction is applied.
- *rrad*(ir) Radius of the real cylindrical boundary.
- ang(ir) Angle of the cylindrical notch. This data is given if and only if the keyword ANG is present. ang(ir) = $\frac{\pi}{2}$ by default (i.e. the correction is applied at every angle).



Figure 8: Cylindrical correction in Cartesian geometry

The only combinations of diagonal symmetry permitted are: X+DIAG Y-DIAG and X-DIAG Y+DIAG. In these cases the geometry must be a square. The only combinations of translational symmetry permitted are: X-TRAN X+ TRAN, Y-TRAN Y+ TRAN and Z-TRAN Z+ TRAN.

The input corresponding to the (descMC) structure are the following:

Table 6: Structure (descMC)

```
 \begin{bmatrix} \texttt{MIX} \{ (imix(i),i=1,lreg) \mid \\ [[\texttt{PLANE } iplan \{ (imix(i),i=1,lp) \mid \texttt{SAME } iplan1 \\ \mid [[\texttt{CROWN} \{ (imix(i),i=1,lc) \mid \texttt{ALL } jmix \mid \texttt{SAME } iplan1 \} ]] \\ \mid [[\texttt{UPTO } ic \texttt{ ALL } jmix \mid \texttt{SAME } iplan1 \} ]] \end{bmatrix}
```

where

MIX	keyword to attribute an material mixture number to each volume inside the axes of sym- metry. When a volume is located inside the axes of symmetry but outside the calculation region it must be declared 'virtual' (for example, the corners of a nuclear reactor). The material mixture number should be specified for each volume before mesh-splitting.
imix	type of material mixture associated with a region. It is important that $imix \leq nmixt$ where $nmixt$ is defined in the module. If $imix=0$, the corresponding volume is replaced by a VOID boundary condition. In this case the volume is considered to be virtual and the flux is not calculated. In the case of a diagonal symmetry, the type indicator must not be specified for the volumes outside the axis of symmetry. These values must be specified in the following order: from X- to X+, from Y- to Y+, from Z- to Z+ and finally radially from the inside out.
PLANE	keyword to attribute mixture numbers to each volume inside a single 2D plane. This option is valid only for 3D geometries, Cartesian or hexagonal.
iplan	plane number for which material mixture are input.
SAME	keyword to attribute the same material mixture numbers of the <i>iplan1</i> plane to the <i>iplan</i> plane. In hexagonal geometry, it can indicate that the mixture numbers of the current crown of the <i>iplanth</i> plane will be identical to those of the same crown of the <i>iplan1th</i> plane.
iplan1	plane number used as reference to input the current plane or crown(s).
lp	number of volumes in a plane. In Cartesian geometry, $lp = lx * ly$ and in hexagonal geometry, $lp = lh$.
CROWN	keyword to attribute mixture numbers to each hexagon of a single crown. This option is only valid for COMPLETE hexagonal geometry definition. Each use of the keyword CROWN increases the crown number by 1. So it is not required to give its number, but crowns must be defined from the center to the peripherical regions of a plane.
lc	number of hexagons in the current crown. For the <i>i</i> th crown of a compelete hexagonal plane, $lc = (i - 1) * 6$. The first crown is composed of only one hexagon.
ALL	keyword to specify that the lc material mixture number of the current crown have the same value $jmix.$
UPTO	keyword to attribute material mixture numbers of the current crown up to the ic one.
ic	number of the last crown in UPTO option. Its value must be greater than equal to the current crown number.

Here we will assume that lreg is the exact number of cells or elementary cases to be considered. For example, if we had used the DIAG option with a geometry of type CAR3D (lx=ly), we would have: lreg=(lx+1)*ly*lz/2.

The following dimensional constraints must also be respected:

- nmerge=number of merged cells (with $nmerge \ge lreg.$),
- ngen=number of generation cells (with $ngen \ge nmerge$.),

The inputs corresponding to the (descPOS) structure are the following:

Table 7: Structure (descPOS)

[MESHX (xxx(i) i=1 lx+1)]
$\begin{bmatrix} \text{MEGHY} (\text{AAA}(1), 1 - 1, 1 + 1) \end{bmatrix}$
$[\operatorname{MESHY}(yyy(1),1=1,1y+1)]$
[MESHZ (zzz(i),i=1,lz+1)]
[RADIUS (rrr(i), i=1, lr+1)]
[SIDE sidhex]
[SPLITX $(ispltx(i),i=1,lx)]$
[SPLITY (isplty(i), i=1, ly)]
[SPLITZ $(ispltz(i),i=1,lz)]$
[SPLITR (<i>ispltr</i> (i),i=1, <i>lr</i>)]

where

MESHX	keyword for the mesh of the geometry along the X axis.
MESHY	keyword for the mesh of the geometry along the Y axis.
MESHZ	keyword for the mesh of the geometry along the Z axis.
RADIUS	keyword for the mesh of the geometry in the radial direction.
SIDE	keyword for the length of a side of a hexagon.
XXX	abscissa, corresponding to the limits of the regions making up the geometry. These values must be given in order, from $X-$ to $X+$. If the geometry presents a diagonal symmetry this data will also be used for the ordinate.
УУУ	ordinate, corresponding to the limits of the regions making up the geometry. These values must be given in order, from $Y-$ to $Y+$.
ZZZ	height, corresponding to the limits of the regions making up the geometry. These values must be given in order, from $Z-$ to $Z+.$
rrr	Radii in the cases of cylindrical (TUBE or TUBEZ), spherical (SPHERE). It is important to note that we must have $rrr(1)=0.0$.
sidhex	length of a side of a hexagon.
SPLITX	keyword for mesh splitting of the geometry along the X axis.
SPLITY	keyword for mesh splitting of the geometry along the Y axis.
SPLITZ	keyword for mesh splitting of the geometry along the Z axis.
SPLITR	keyword for mesh splitting of the geometry in the radial direction.

ispltx	number of sub-volumes that will be defined for each row of the volume along the X-axis. If the geometry presents a diagonal symmetry this input will also be used for the splitting along the Y-axis. By default, $ispltx=1$.
isplty	number of sub-volumes that will be defined for each row of the volume along the Y-axis. If the geometry presents a diagonal symmetry this input will also be used for the splitting along the X-axis. By default, $isplty=1$.
ispltz	number of sub-volumes that will be defined for each row of the volume along the Z-axis. By default, $ispltz(i)=1$.
ispltr	the value of <i>ispltr</i> gives the number of sub-volumes that will be defined for each tube or each spherical shell. A negative value permits a splitting into equal sub-volumes; a positive value permits a splitting into equal sub-radius spacings. By default, $ispltr=1$.

The user of the options described above should take care not to exceed the limits imposed by the amount of dynamically allocated memory available. For a pure geometry, let us define the variables lxp, lyp, lzp and lrp as:

$$lxp = \sum_{i=1}^{lx} ispltx(i)$$
$$lyp = \sum_{i=1}^{ly} isplty(i)$$
$$lzp = \sum_{i=1}^{lz} ispltz(i)$$
$$lrp = \sum_{i=1}^{lr} ispltr(i)$$

thus, the limits that must be respected are the following:

- $lxp \ge maxpts$ for a CAR1D geometry.
- $lh \ge maxpts$ for a HEX geometry.
- $lrp \geq maxpts$ for the TUBE and SPHERE geometries.
- $lxp * lyp \ge maxpts$ for the CAR2D geometry without diagonal symmetry.
- $lxp * (lyp + 1)/2 \ge maxpts$ for the CAR2D geometry with diagonal symmetry.
- $lrp * lzp \ge maxpts$ for the TUBEZ geometry.
- $lxp * lyp * lzp \ge maxpts$ for the CAR3D geometry without diagonal symmetry.
- $lxp * (lyp + 1) * lzp/2 \ge maxpts$ for the CAR3D geometry with diagonal symmetry.
- $lh*lzp \ge maxpts$ for the HEXZ geometry.

1.3.2 Examples of geometries

We will now give a few examples which will permit users to better understand the procedure used to define the geometries in TRIVAC.

```
1. Slab geometry (see Fig. 9):

GEOMETRY1 := GEO: :: CAR1D 6

X- VOID X+ ALBE 1.2

MESHX 0.0 0.1 0.3 0.5 0.6 0.8 1.0

SPLITX 2 2 2 1 2 1

MIX 1 2 3 4 5 6

;

\beta=0.0 \beta=0.0 \beta=0.1 \beta=0.2 \beta=1.2 \beta=1.2 \beta=1.2 \beta=1.2 \beta=0.1 \beta=0.1
```

Figure 9: Slab geometry with mesh-splitting

2. Two-dimensional hexagonal geometry (see Fig. 10):

```
GEOMETRY4 := GEO: :: HEX 12
HBC S30 ALBE 1.6
SIDE 1.3
MIX 1 1 1 2 2 2 3 3 3 4 5 6
;
```



Figure 10: Two-dimensional hexagonal geometry

1.4 The MAC: module

In TRIVAC the macroscopic cross sections and diffusion coefficients are read from the input data file using REDLEC. The general format of the data for the MAC: module in TRIVAC is the following:

Table 8: Structure (MAC:)

$MACR1 := MAC: [\{ MACR1 | MACR2 \}] :: (mac_data)$

where

- MACR1 character*12 name of the LCM object (type L_MACROLIB) containing the new Macrolib produced by the module. A Macrolib contains macroscopic cross sections and diffusion coefficients. If MACR1 appears on both LHS and RHS, it is updated; otherwise, it is created. If MACR1 is created, all macroscopic cross sections and diffusion coefficients are first initialized to zero.
- MACR2 character*12 name of the LCM object (type L_MACROLIB) containing a read-only Macrolib. The information existing in MACR2 is copied into MACR1, but MACR2 is not modified.

(mac_data) structure containing the data to module MAC: (see Sect. 1.4.1).

1.4.1 Data input for module MAC:

Table 9: Structure (mac_data)

```
[ EDIT iprint ]
[ NGRO ngroup ]
[ NIFI nifiss ]
[ DELP ndel ]
[ ANIS naniso ]
[ NMIX nmixt ]
[ DELP ndg ]
[ ANIS naniso ]
[ ALBP nalbp (albedp(ia),ia=1,nalbp) ]
[ READ INPUT { [[ (macxs) ]] | OLD (triv2) | DOLD (trip2) } ]
[[ STEP istep READ INPUT [[ (macxs) ]] ]]
```

where

EDIT keyword used to set *iprint*.

iprint index used to control the printing in module MAC:. =0 for no print. The macroscopic cross sections will be printed if the parameter *iprint* is greater than or equal to 2. The transfer cross sections will be printed if this parameter is greater than or equal to 3.

NGRO	keyword used to define the number of energy groups. This data is given if and only if $MACR1$ is created.
ngroup	the number of energy groups used for the calculations in TRIVAC.
NIFI	keyword used to specify the maximum number of fissile spectrum associated with each mixture. Each fission spectrum generally represents a fissile isotope. This information is required only if <i>MACLIB</i> is created and the cross sections are taken directly from the input data stream.
nifiss	the maximum number of fissile isotopes per mixture. The default value is $nifiss=1$.
DELP	keyword used to specify the number of delayed neutron groups.
ndel	the number of delayed neutron groups. The default value is $ndel=0$.
ANIS	keyword used to specify the maximum level of anisotropy permitted in the scattering cross sections. This information is required only if <i>MACLIB</i> is created and the cross sections are taken directly from the input data stream.
naniso	number of Legendre orders for the representation of the scattering cross sections. The default value is <i>naniso</i> =1 corresponding to the use of isotropic scattering cross sections.
NMIX	keyword used to define the number of material mixtures. This data is given if and only if $MACR1$ is created.
nmixt	the maximum number of material mixtures (a material mixture is characterized by a dis- tinct set of macroscopic cross sections).
DELP	keyword used to set ndg . This data is used only if the fission spectrum $\vec{\chi}_p$ is different from the delayed neutron spectrum $\vec{\chi}_i$ for each precursor group i .
ndg	number of delayed neutron groups.
ANIS	keyword used to specify the maximum level of anisotropy permitted in the diffusion cross sections. This data is given only if $MACR1$ is created.
naniso	the maximum level of anisotropy. The default value is $naniso=1$.
ALBP	keyword used for the input of the physical albedos.
nalbp	the maximum number of physical albedos.
albedp	physical albedos (real numbers).
STEP	keyword used to create a perturbation directory.
istep	the index of the perturbation directory.
READ	keyword used to specify input of the cross section information from default input by REDLEC.
(\max)	structure describing the format used for reading the mixture cross sections and diffusion coefficients (or perturbation values of the cross sections and diffusion coefficients) from the input data file.
OLD	keyword used to specify input of the cross section information from default input by REDLEC in the TRIVAC-2 format. The nuclear data will be translated into TRIVAC format and printed on the listing.
(triv2)	structure describing the format used for reading the mixture cross sections and diffusion coefficients from the input data file in TRIVAC-2 format.

IGE-293

- DOLDkeyword used to specify perturbed input of the cross section information from default input
by REDLEC in the TRIVAC-2 format. The perturbed nuclear data will be translated into
TRIVAC format and printed on the listing.
- (trip2) structure describing the format used for reading the mixture values of the perturbed cross sections and diffusion coefficients from the input data file in TRIVAC-2 format.

1.4.2 Description of the nuclear data

Table 10: Structure (macxs)

MIX matnum
[{ NTOTO TOTAL } (xssigt(jg), jg=1,ngroup)]
[NTOT1 (xssig1(jg), jg=1,ngroup)]
[TRANC (xsstra(jg), jg=1,ngroup)]
[NUSIGF $((xssigf(jf,jg), jg=1,ngroup), jf=1,nifiss)]$
[CHI ((xschi(jf,jg), jg=1,ngroup), jf=1,nifiss)]
[FIXE (xsfixe(jg), jg=1,ngroup)]
[DIFF (diff(jg), jg=1,ngroup)]
[DIFFX (xdiffx(jg), jg=1,ngroup)]
[DIFFY (xdiffy(jg), jg=1,ngroup)]
[DIFFZ (xdiffz(jg), jg=1,ngroup)]
[NUSIGD (((xssigd(jf,idel,jg), jg=1,ngroup), idel=1,ndel), jf=1,nifiss) $]$
[CHDL (((xschid(jf,idel,jg), jg=1,ngroup), idel=1,ndel), jf=1,nifiss)]
[OVERV (overv(jg), jg=1, ngroup)]
[H-FACTOR (xhfact(jg), jg=1, ngroup)]
[SCAT((nbscat(jl,jg), ilastg(jl,jg), (scat(jl,jg,ig), ig=1, nbscat(jl,jg)), jg=1, ngroup), jl=1, naniso)]

where

MIX	keyword to specify that the macroscopic cross sections associated with a new mixture are to be read.
matnum	identifier for the next mixture to be read. The maximum value permitted for this identifier is <i>nmixt</i> . When <i>matnum</i> is absent, the mixtures are numbered consecutively starting with 1 or with the last mixture number read either on the GOXS or the input stream.
NTOTO	keyword to specify that the total macroscopic cross sections for this mixture follows.
TOTAL	alias keyword for NTOTO.
xssigt	array representing the multigroup total macroscopic cross section (Σ^g in cm ⁻¹) associated with this mixture.
NTOT1	keyword to specify that the $P_1\mbox{-weighted}$ total macroscopic cross sections for this mixture follows.
xssig1	array representing the multigroup P_1 -weighted total macroscopic cross section (Σ_1^g in cm ⁻¹) associated with this mixture.
TRANC	keyword to specify that the transport correction macroscopic cross sections for this mixture follows.

xsstra	array representing the multigroup transport correction macroscopic cross section (Σ_{tc}^{g} in cm ⁻¹) associated with this mixture.
NUSIGF	keyword to specify that the macroscopic fission cross section multiplied by the average number of neutrons per fission for this mixture follows.
xssigf	array representing the multigroup macroscopic fission cross section multiplied by the average number of neutrons per fission $(\nu \Sigma_f^g \text{ in } \text{cm}^{-1})$ for all the fissile isotopes associated with this mixture.
CHI	keyword to specify that the fission spectrum for this mixture follows.
xschi	array representing the multigroup fission spectrum (χ^g) for all the fissile isotopes associated with this mixture.
FIXE	keyword to specify that the fixed neutron source density for this mixture follows.
xsfixe	array representing the multigroup fixed neutron source density for this mixture $(S^g \text{ in } s^{-1}cm^{-3})$.
DIFF	keyword to specify that the isotropic diffusion coefficient for this mixture follows.
diff	array representing the multigroup isotropic diffusion coefficient for this mixture $(D^g$ in $cm)$.
DIFFX	keyword for input of the X -directed diffusion coefficient.
xdiffx	array representing the multigroup X–directed diffusion coefficient $(D^g_x \mbox{ in cm})$ for the mixture matnum.
DIFFY	keyword for input of the Y -directed diffusion coefficient.
xdiffy	array representing the multigroup Y–directed diffusion coefficient $(D_y^g \mbox{ in cm})$ for the mixture matnum.
DIFFZ	keyword for input of the Z -directed diffusion coefficient.
xdiffz	array representing the multigroup Z–directed diffusion coefficient $(D^g_z \mbox{ in cm})$ for the mixture matnum.
NUSIGD	keyword to specify that the delayed macroscopic fission cross section multiplied by the average number of neutrons per fission for this mixture follows.
xssigd	array representing the delayed multigroup macroscopic fission cross section multiplied by the average number of neutrons per fission $(\nu \Sigma_f^{g,idel}$ in cm ⁻¹) for all the fissile isotopes associated with this mixture.
CHDL	keyword to specify that the delayed fission spectrum for this mixture follows.
xschid	array representing the delayed multigroup fission spectrum $(\chi^{g,idel})$ for all the fissile isotopes associated with this mixture.
OVERV	keyword for input of the multigroup average of the inverse neutron velocity.
overv	array representing the multigroup average of the inverse neutron velocity $(<1/v>_m^g)$ for the mixture matnum.
H-FACTOR	keyword to specify that the power factor for this mixture follows.
hfact	array representing the multigroup power factor for this mixture $(H^g \text{ in } MeV \ cm^{-1})$.
SCAT	keyword to specify that the macroscopic scattering cross section matrix for this mixture follows.

- *nbscat* array representing the number of secondary groups ig with non vanishing macroscopic scattering cross section towards the primary group jg considered for each anisotropy level associated with this mixture.
- *ilastg* array representing the group index of the most thermal group with non-vanishing macroscopic scattering cross section towards the primary group jg considered for each anisotropy level associated with this mixture.
- xsscat array representing the multigroup macroscopic scattering cross section $(\Sigma_{sl}^{ig \to jg} \text{ in cm}^{-1})$ from the secondary group ig towards the primary group jg considered for each anisotropy level associated with this mixture. The elements are ordered using decreasing secondary group number ig, from *ilastg* to (*ilastg-nbscat*+1), and an increasing primary group number jg.

For example, the two group isotropic and linearly anisotropic scattering cross sections (ngroup=2, naniso=2) given by:

L	$\Sigma_{s,l}^{1 \to 1}$	$\Sigma_{s,l}^{1 \to 2}$	$\Sigma_{s,l}^{2 \to 1}$	$\Sigma_{s,l}^{2 \to 2}$
0	$0.50 {\rm ~cm^{-1}}$	$0.20 {\rm ~cm^{-1}}$	$0.03 {\rm ~cm^{-1}}$	$0.40 {\rm ~cm^{-1}}$
1	$0.05 \ {\rm cm}^{-1}$	$0.00 \ {\rm cm}^{-1}$	$0.00 \ {\rm cm}^{-1}$	$0.04 \ {\rm cm}^{-1}$

must be entered as:

SCAT (*L=0*) 2 2 (*2->1*) 0.03 (*1->1*) 0.50 2 2 (*2->2*) 0.40 (*1->2*) 0.20 (*L=1*) 1 1 (*1->1*) 0.05 1 2 (*2->2*) 0.04

1.5 The BIVACT: module

The BIVACT: module is used to perform a BIVAC-type TRACKING on a 1D/2D geometry.^[3,4,14] The geometry is analyzed and a LCM object with signature L_BIVAC is created with the following information:

- Diagonal and hexagonal symmetries are unfolded and the mesh-splitting operations are performed. Volumes, material mixture and averaged flux recovery indices are computed on the resulting geometry.
- A finite element discretization is performed and the corresponding numbering is saved.
- The unit finite element matrices (mass, stiffness, etc.) are recovered.

The calling specifications are:

Table 11: Structure (BIVACT:)

TRACK := BIVACT: [TRACK] GEOM :: (bivact_data)

where

TRACK character*12 name of the LCM object (type L_BIVAC) containing the TRACKING information. If TRACK appears on the RHS, the previous settings will be applied by default.

GEOM character*12 name of the LCM object (type L_GEOM) containing the geometry.

(bivact_data) structure containing the data to module BIVACT: (see Sect. 1.5.1).

1.5.1 Data input for module BIVACT:

Table 12: Structure (bivact_data)

where

EDIT keyword used to set *iprint*.

iprint index used to control the printing in module BIVACT:. =0 for no print; =1 for minimum printing (default value); Larger values produce increasing amounts of output.

TITL	keyword which allows the run title to be set.
TITLE	the title associated with a TRIVAC run. This title may contain up to 72 characters. The default when TITL is not specified is no title.
MAXR	keyword which permits the maximum number of regions to be considered during a TRIVAC run to be specified.
maxpts	maximum dimensions of the problem to be considered. The default value is set to the number of regions previously computed by the GEO: module but this value is insufficient if symmetries or mesh-splitting are specified.
PRIM	keyword to set a primal finite element (classical) discretization.
DUAL	keyword to set a mixed-dual finite element discretization. If the geometry is hexagonal, a Thomas-Raviart-Schneider method is used. ^[15]
MCFD	keyword to set a mesh-centered finite difference discretization in hexagonal geometry.
ielem	order of the finite element representation. The values permitted are 1 (linear polynomials), 2 (parabolic polynomials), 3 (cubic polynomials) or 4 (quartic polynomials). By default <i>ielem</i> =1.
icol	type of quadrature used to integrate the mass matrices. The values permitted are 1 (an- alytical integration), 2 (Gauss-Lobatto quadrature) or 3 (Gauss-Legendre quadrature). By default <i>icol=2</i> . The analytical integration corresponds to classical finite elements; the Gauss-Lobatto quadrature corresponds to a variational or nodal type collocation and the Gauss-Legendre quadrature corresponds to superconvergent finite elements.
isplh	type of hexagonal mesh-splitting. This data is given only if the geometry is 2D hexagonal. The values permitted are 1 (full hexagons), 2 for splitting each hexagon into 6 triangles, 3 for splitting each hexagon into 24 triangles, 5 for splitting each hexagon into 96 triangles, 9 for splitting each hexagon into 384 triangles and 17 for splitting each hexagon into 1536 triangles. The values permitted with the Thomas-Raviart-Schneider method are: 1 (3 lozanges per hexagon), > 1 for performing a mesh-splitting in $3 \times isplh^2$ losanges per hexagon.
PN	keyword to set a spherical harmonics (P_n) expansion of the flux. ^[9]
SPN	keyword to set a simplified spherical harmonics (SP_n) expansion of the flux. ^[9,10] This option is currently available with 1D and 2D Cartesian geometries and with 2D hexagonal geometries.
n	order of the P_n or SP_n expansion (odd number). Set to zero for diffusion theory (default value).
SCAT	keyword to limit the anisotropy of scattering sources.
DIFF	keyword to force using $1/3D^g$ as Σ_1^g cross sections. A P_1 or SP_1 method will therefore behave as diffusion theory.
iscat	number of terms in the scattering sources. $iscat = 1$ is used for isotropic scattering in the laboratory system. $iscat = 2$ is used for linearly anisotropic scattering in the laboratory system. The default value is set to $n + 1$ in P_n or SP_n case.
VOID	keyword to set the number of base points in the Gauss-Legendre quadrature used to integrate void boundary conditions if $icol = 3$ and $n \neq 0$.
nvd	type of quadrature. The values permitted are: 0 (use a $(n+2)$ -point quadrature consistent with P_n theory), 1 (use a $(n+1)$ -point quadrature consistent with S_{n+1} theory), 2 (use an analytical integration of the void boundary conditions). By default $nvd=0$.

Various finite element approximations can be obtained by combining different values of *ielem* and *icol*:

- PRIM 1 1 : Linear finite elements;
- PRIM 1 2 : Mesh corner finite differences;
- PRIM 1 3 : Linear superconvergent finite elements;
- PRIM 2 1 : Quadratic finite elements;
- PRIM 2 2 : Quadratic variational collocation method;
- PRIM 2 3 : Quadratic superconvergent finite elements;
- PRIM 3 1 : Cubic finite elements;
- PRIM 3 2 : Cubic variational collocation method;
- PRIM 3 3 : Cubic superconvergent finite elements;
- PRIM 4 2 : Quartic variational collocation method;
- DUAL 1 1 : Mixed-dual linear finite elements;
- DUAL 1 2 : Mesh centered finite differences;
- DUAL 1 3 : Mixed-dual linear superconvergent finite elements (numerically equivalent to PRIM 1 3);
- DUAL 2 1 : Mixed-dual quadratic finite elements;
- DUAL 2 2 : Quadratic nodal collocation method;
- DUAL 2 3 : Mixed-dual quadratic superconvergent finite elements (numerically equivalent to PRIM 2 3);
- DUAL 3 1 : Mixed-dual cubic finite elements;
- DUAL 3 2 : Cubic nodal collocation method;
- DUAL 3 3 : Mixed-dual cubic superconvergent finite elements (numerically equivalent to PRIM 3 3);
- DUAL 4 2 : Quartic nodal collocation method;

1.6 The TRIVAT: module

The TRIVAT: module is used to perform a TRIVAC-type TRACKING on a 1D/2D/3D geometry.^[4-8, 14] The geometry is analyzed and a LCM object with signature L_TRIVAC is created with the following information:

- Diagonal and hexagonal symmetries are unfolded and the mesh-splitting operations are performed. Volumes, material mixture and averaged flux recovery indices are computed on the resulting geometry.
- A finite element discretization is performed and the corresponding numbering is saved.
- The unit finite element matrices (mass, stiffness, etc.) are recovered.
- Indices related to an ADI preconditioning with or without supervectorization are saved.

The calling specifications are:

Table 13: Structure (TRIVAT:)

TRACK := TRIVAT: [TRACK] GEOM :: (trivat_data)

where

- TRACK character*12 of the LCM object (type L_TRIVAC) containing the TRACKING information. If TRACK appears on the RHS, the previous settings will be applied by default.
- GEOM character*12 of the LCM object (type L_GEOM) containing the geometry.

(trivat_data) structure containing the data to module TRIVAT: (see Sect. 1.6.1).

1.6.1 Data input for module TRIVAT:

Table 14: Structure (trivat_data)

```
[ EDIT iprint ]
[ TITL TITLE ]
[ MAXR maxpts ]
[ { PRIM [ ielem [ isplh ] ] | DUAL [ ielem icol [ isplh ] ] | MCFD [ ielem [ isplh ] ] | LUMP [ ielem ] } ]
[ SPN n [ SCAT [ DIFF ] iscat ] [ VOID nvd ] ]
[ ADI nadi ]
[ VECT [ iseg ] [ PRTV impv ] ]
;
```

where

EDIT keyword used to set *iprint*.

iprint	index used to control the printing in module TRIVAT:. =0 for no print; =1 for minimum printing (default value); Larger values produce increasing amounts of output.
TITL	keyword which allows the run title to be set.
TITLE	the title associated with a TRIVAC run. This title may contain up to 72 characters. The default when TITL is not specified is no title.
MAXR	keyword which permits the maximum number of regions to be considered during a TRIVAC run to be specified.
maxpts	maximum dimensions of the problem to be considered. The default value is set to the number of regions previously computed by the GEO: module but this value is insufficient if symmetries or mesh-splitting are specified.
PRIM	keyword to set a discretization based on the variational collocation method.
DUAL	keyword to set a mixed-dual finite element discretization. If the geometry is hexagonal, a Thomas-Raviart-Schneider method is used. ^[15]
MCFD	keyword to set a discretization based on the nodal collocation method. The mesh centered finite difference approximation is the default option and is generally set using MCFD 1. The MCFD approximations are numerically equivalent to the DUAL approximations with $icol=2$; however, the MCFD approximations are less expensive.
LUMP	keyword to set a discretization based on the nodal collocation method with serendipity approximation. The serendipity approximation is different from the MCFD option in cases with $ielem \geq 2$. This option is not available for hexagonal geometries.
ielem	order of the finite element representation. The values permitted are: 1 (linear polynomials), 2 (parabolic polynomials), 3 (cubic polynomials) or 4 (quartic polynomials). By default <i>ielem</i> =1.
icol	type of quadrature used to integrate the mass matrices. The values permitted are: 1 (analytical integration), 2 (Gauss-Lobatto quadrature) or 3 (Gauss-Legendre quadrature). By default <i>icol</i> =2. The analytical integration corresponds to classical finite elements; the Gauss-Lobatto quadrature corresponds to a variational or nodal type collocation and the Gauss-Legendre quadrature corresponds to superconvergent finite elements.
isplh	type of hexagonal mesh-splitting. This data is given only if the geometry is 2D or 3D hexagonal. The values permitted with the MCFD option are: 1 (full hexagons), 2 for splitting each hexagon into 6 triangles, 3 for splitting each hexagon into 24 triangles, etc. The values permitted with the PRIM option are: 1 (full hexagons) and 2 for splitting each hexagon into 6 triangles. The values permitted with the Thomas-Raviart-Schneider method are: 1 (3 lozenges per hexagon), > 1 for performing a mesh-splitting in $3 \times isplh^2$ losanges per hexagon.
SPN	keyword to set a simplified spherical harmonics (SP_n) expansion of the flux. ^[9,10] This option is available with 1D, 2D and 3D Cartesian geometries and with 2D and 3D hexagonal geometries.
n	order of the P_n or SP_n expansion (odd number). Set to zero for diffusion theory (default value).
SCAT	keyword to limit the anisotropy of scattering sources.
DIFF	keyword to force using $1/3D^g$ as Σ_1^g cross sections. A P_1 or SP_1 method will therefore behave as diffusion theory.
iscat	number of terms in the scattering sources. $iscat = 1$ is used for isotropic scattering in the laboratory system. $iscat = 2$ is used for linearly anisotropic scattering in the laboratory system. The default value is set to $n + 1$ in P_n or SP_n case.

VOID	keyword to set the number of base points in the Gauss-Legendre quadrature used to integrate void boundary conditions if $icol = 3$ and $n \neq 0$.
nvd	type of quadrature. The values permitted are: 0 (use a $(n+2)$ -point quadrature consistent with P_n theory), 1 (use a $(n+1)$ -point quadrature consistent with S_{n+1} theory), 2 (use an analytical integration of the void boundary conditions). By default $nvd=0$.
ADI	keyword to set the number of ADI iterations at the inner iterative level.
nadi	number of ADI iterations (default: $nadi = 2$).
VECT	keyword to set an ADI preconditionning with supervectorization. By default, TRIVAC uses an ADI preconditionning without supervectorization.
iseg	width of a vectorial register. <i>iseg</i> is generally a multiple of 64. By default, <i>iseg</i> =64.
PRTV	keyword used to set <i>impv</i> .
impv	index used to control the printing in supervectorization subroutines. =0 for no print; =1 for minimum printing (default value); Larger values produce increasing amounts of output.

Various finite element approximations can be obtained by combining different values of *ielem* and *isplh* (see Sect. 1.5).

1.7 The BIVACA: module

The BIVACA: module is used to compute the finite element system matrices (type L_SYSTEM) corresponding to a BIVAC TRACKING (type L_BIVAC) and to a set of nuclear properties (type L_MACROLIB). The calling specifications are:

Table 15: Structure (BIVACA:)

SYST := BIVACA: [SYST] MACRO TRACK :: (bivaca_data)

where

SYST	<code>character*12</code> name of the LCM object (type <code>L_SYSTEM</code>) containing the system matrices. If $SYST$ appears on the RHS, the system matrices previously stored in $SYST$ are kept.
MACRO	$\label{eq:character*12} character*12 \ \text{name of the LCM object} \ (type \ \texttt{L_MACROLIB}) \ containing \ the \ macroscopic \ cross \ sections \ and \ diffusion \ coefficients.$
TRACK	$\texttt{character*12} \text{ name of the LCM object (type L_BIVAC) containing the BIVAC TRACKING.}$
(bivaca_data	a) structure containing the data to module BIVACA: (see Sect. 1.7.1).

1.7.1 Data input for module BIVACA:

Table 16: Structure (bivaca_data)

[EDIT *iprint*] [UNIT]

where

:

EDIT keyword used to set *iprint*.

- *iprint* index used to control the printing in module BIVACA:. =0 for no print; =1 for minimum printing (default value); Larger values produce increasing amounts of output.
- UNIT A system matrix corresponding to cross sections all set to 1.0 is computed. This keyword is mandatory if the system matrices in *SYST* are going to be used by INIKIN: or KINSOL: modules (see Sects. 1.14 and 1.15).

1.8 The TRIVAA: module

The TRIVAA: module is used to compute the finite element system matrices (type L_SYSTEM) corresponding to a TRIVAC TRACKING (type L_TRIVAC) and to a set of nuclear properties (type L_MACROLIB). The calling specifications are:

Table 17: Structure (TRIVAA:)

SYST := TRIVAA: [SYST] MACRO TRACK [DMACRO] :: (trivaa_data)

where

SYST	<code>character*12</code> name of the LCM object (type <code>L_SYSTEM</code>) containing the system matrices. If $SYST$ appears on the RHS, the system matrices previously stored in $SYST$ are kept.
MACRO	$\label{eq:character*12} character*12 \ name of the \ \mbox{LCM} object \ (type \ \mbox{L_MACROLIB}) \ containing the \ macroscopic \ cross sections \ and \ diffusion \ coefficients.$
TRACK	$\texttt{character*12} \text{ name of the LCM object (type L_TRIVAC) containing the TRIVAC TRACKING.}$
DMACRO	$\label{eq:character*12} character*12 name of the LCM object (type L_MACROLIB) containing derivatives or perturbations of the macroscopic cross sections and diffusion coefficients. If DMACRO is given, only the derivatives or perturbations of the system matrices are computed.$

(trivaa_data) structure containing the data to module TRIVAA: (see Sect. 1.8.1).

1.8.1 Data input for module TRIVAA:

Table 18: Structure (trivaa_data)

[EDIT iprint] [SKIP] [{ DERI | PERT }] [UNIT][OVEL]

where	
EDIT	keyword used to set <i>iprint</i> .
iprint	index used to control the printing in module TRIVAA:. =0 for no print; =1 for minimum printing (default value); Larger values produce increasing amounts of output.
SKIP	keyword used to skip the system matrix assembly but to perform the $L - D - L^T$ factor- ization. Use the system matrices already present in SYST.
DERI	The information recovered from <i>DMACRO</i> is used as derivatives of nuclear properties with respect to a state variable. Derivatives of system matrices with respect to the same state variable are computed.

- UNIT A system matrix corresponding to cross sections all set to 1.0 is computed. This keyword is mandatory if the system matrices in *SYST* are going to be used by INIKIN: or KINSOL: modules (see Sects. 1.14 and 1.15).
- **OVEL** The reciprocal neutron velocities for each material mixture are recovered from the input MACROLIB *MACRO* and used to compute the corresponding system matrices. This capability is deprecated.

1.9 The FLUD: module

The FLUD: module is used to compute the solution to an eigenvalue problem corresponding to a set of system matrices (type L_SYSTEM). The calling specifications are:

Table 19: Structure (FLUD:)

FLUX := FLUD: [FLUX] SYST TRACK [MACRO] :: (flud_data)

where

FLUX	character*12 name of the LCM object (type L_FLUX) containing the solution. If FLUX
	appears on the RHS, the solution previously stored in FLUX is used to initialize the new
	iterative process; otherwise, a uniform unknown vector is used.

- SYST character*12 name of the LCM object (type L_SYSTEM) containing the system matrices.
- TRACK character*12 name of the LCM object (type L_TRACK) containing the TRACKING.
- MACRO character*12 name of the optional LCM object (type L_MACROLIB) containing the cross sections. This object is only used to set a link to the MACROLIB name inside the FLUX object. By default, the name of the MACROLIB is recovered from the link in the SYSTEM object.
- (flud_data) structure containing the data to module FLUD: (see Sect. 1.9.1).

1.9.1 Data input for module FLUD:

Table 20: Structure (flud_data)

```
[ EDIT iprint ]
[ { VAR1 | ACCE } icl1 icl2 ]
[ EXTE [ maxout ] [ epsout ] ]
[ THER [ maxthr ] [ epsthr ] ]
[ ADI nadi ]
[ ADJ ]
[ MONI lmod [ RAND ] ]
[ RELAX relax ]
;
```

where

EDIT keyword used to set *iprint*.

iprint index used to control the printing in module FLUD: =0 for no print; =1 for minimum printing (default value); =2 iteration history is printed; =3 the solution is printed; =4 at each iteration, the new solution is compared to a reference solution previously stored in *FLUX* under name REF; =5 the convergence histogram is stored in *FLUX*.

VAR1	keyword used to set the parameters ((icl1 and icl2)	of the symmetrical	variational acceler-
	ation technique (SVAT).			

ACCE alias keyword for VAR1.

- *icl1* number of free outer iterations in a cycle of the SVAT. The default value is icl1 = 3.
- *icl2* number of accelerated outer iterations in a cycle of the SVAT. The default value is icl2 = 3. A convergence in free iterations is obtained by setting icl1 = 200 (or icl1 = maxx0) and icl2 = 0.
- **EXTE** keyword to specify that the control parameters for the external iteration are to be modified.
- maxout maximum number of external iterations. The fixed default value is maxout = 200.
- epsout convergence criterion for the external iterations. The fixed default value is $epsout = 1.0 \times 10^{-4}$. The outer iterations are stopped when the following criteria is reached:

$$\max_{i} |\Phi_i^{(k-1)} - \Phi_i^{(k)}| \leq epsout \times \max_{i} |\Phi_i^{(k)}|$$

where $\vec{\Phi}^{(k)} = \operatorname{col}\{\Phi_i^{(k)}; i = 1, I\}$ is the product of the *B* matrix times the unknown vector at the *k*-th outer iteration.

THER keyword to specify that the control parameters for the thermal iterations are to be modified.

- maxthr maximum number of thermal iterations. The fixed default value is <math>maxthr = 0 corresponding to no thermal iterations.
- epsthr convergence criterion for the thermal iterations. The fixed default value is $epsthr = 1.0 \times 10^{-2}$.
- ADI keyword used to set *nadi* in cases where Trivac is used.
- nadi number of alternating direction implicit (ADI) inner iterations per outer iteration. The default value is nadi = 1. If this value causes a failure of the acceleration process, it is recommended that a larger value be tried. The optimal choice is generally the minimum value of nadi which allows a convergence in less than 75 outer iterations. nadi = 1 or nadi = 2 is generally the best choice for production-type calculations. The greater nadi is, the smaller the asymptotic convergence constant (ACC) becomes. Taking an arbitrary large value (e.g., nadi = 20) leads to numerical results identical to those of the inverse power method where the system matrices are accurately inverted at each outer iteration (at a prohibitive CPU cost). In this case, the ACC is almost equal to the dominance ratio of the iterative matrix. The default value is recovered in the state vector of the TRACKING object TRACK.
- ADJ keyword used to obtain the solution to both the direct and adjoint eigenvalue problems. The adjoint solution is required if we subsequently want to perform a perturbation calculation.
- MONI keyword used to obtain the first harmonics of the solution and to set *lmod*. A full core representation of the reactor should be used to compute its harmonics. If symmetries are set in the geometry, some harmonics may be skipped. If the reactor is symmetric, a uniform initial estimate of the harmonics may cause some harmonics to be skipped; the keyword RAND should therefore be used.
- *lmod* the *lmod* first bi-orthonormalized harmonics of the solution are computed using the SVATaccelerated preconditioned power method with a Hotelling deflation procedure.^[11]
- RAND keyword used to initialize the harmonics calculations (option MONI) with a random estimate rather than a uniform estimate. This option has no effect if *FLUX* appears on the RHS.

- **RELAX** keyword used to set the relaxation parameter. This keyword must be specified each time a relaxation is required.
- relax relaxation parameter selected in the interval $0 < relax \leq 1.0$ and used to update the flux information in the *FLUX* object. The updated value is taken equal to (1.0-relax) times the previous value (given in the RHS *FLUX* object) plus relax times the value computed within current FLUD: call. The default value is relax = 1.0.

1.10 The DELTA: module

The DELTA: module is used to compute the source components of a fixed source eigenvalue problem corresponding to a set of unperturbed and perturbation system matrices (type L_SYSTEM).

In the direct case, the fixed source is computed as:

$$\vec{S} = (\delta \mathbb{A} - \lambda_o \, \delta \mathbb{B}) \, \vec{\Phi} - \delta \lambda \, \mathbb{B}_o \, \vec{\Phi} \tag{1.1}$$

where the direct source vector \vec{S} is orthogonal to the unperturbed adjoint flux Φ^* . In the adjoint case, the fixed source is computed as:

 $ec{S}^{*} = \left(\delta \ \mathbb{A}^{ op} - \lambda_{
ho} \ \delta \mathbb{B}^{ op}
ight) ec{\Phi}^{*} - \delta \lambda \ \mathbb{B}_{
ho}^{ op} \ ec{\Phi}^{*}$

where the adjoint source vector \vec{S}^* is orthogonal to the unperturbed direct flux Φ and where $\delta \lambda$ is the perturbation of the eigenvalue, as computed from the Rayleigh ratio.

The calling specifications are:

Table 21: Structure (DELTA:)

GPT := DELTA: [GPT] FLUX0 SYST0 DSYST TRACK :: (delta_data)

where

GPT	character*12 name of the LCM object (type L_GPT) containing the fixed source. If GPT appears on the RHS, this information is used to initialize the state vector.
FLUX0	$\texttt{character*12} \text{ name of the LCM object (type L_FLUX) containing the unperturbed flux.}$
SYST0	$\tt character*12$ name of the LCM object (type <code>L_SYSTEM</code>) containing the unperturbed system matrices.
DSYST	$\tt character*12$ name of the LCM object (type <code>L_SYSTEM</code>) containing a perturbation to the system matrices.
TRACK	$character*12$ name of the LCM object (type L_TRACK) containing the TRACKING.
(delta_data)	structure containing the data to module DELTA: (see Sect. 1.10.1).

1.10.1 Data input for module DELTA:

Table 22: Structure (delta_data)

[EDIT *iprint*] [ADJ] .

where

(1.2)

IGE-293

EDIT k	eyword	used	to	set	iprint.
--------	--------	------	----	----------------------	---------

iprint index used to control the printing in module DELTA:.

ADJ keyword used to set the source on an adjoint fixed source eigenvalue problem.

1.11 The GPTFLU: module

The GPTFLU: module is used to compute the solution to a fixed source eigenvalue problem corresponding to a set of unperturbed system matrices and sources vectors.

If \vec{S} is the source term of the explicit generalized adjoint equation, this module will solve:

$$\left(\mathbb{A}_{o} - \lambda_{o} \,\mathbb{B}_{o}\right)\vec{\Gamma}_{i} = \vec{S}_{i} \tag{1.3}$$

where the direct source vector \vec{S}_i is orthogonal to the adjoint flux.

If \vec{S} is the source term of the implicit generalized adjoint equation, this module will solve:

$$\left(\mathbb{A}_{o}^{\top} - \lambda_{o} \,\mathbb{B}_{o}^{\top}\right) \,\widetilde{\Gamma}_{j}^{*} = \widetilde{S}_{j}^{*} \tag{1.4}$$

where the adjoint source vector \vec{S}_{i}^{*} is orthogonal to the direct flux.

The calling specifications are:

Table 23: Structure (GPTFLU:)

FLUX_GPT := GPTFLU: [FLUX_GPT] GPT FLUX0 SYST TRACK :: (gptflu_data)

where

FLUX_GPT	character*12 name of the LCM object (type L_FLUX) containing the GPT solution. If $FLUX_GPT$ appears on the RHS, the solution previously stored in $FLUX_GPT$ is used to initialize the new iterative process; otherwise, a uniform unknown vector is used.
GPT	<code>character*12</code> name of the LCM object (type <code>L_GPT</code>) containing the fixed sources.
FLUX0	$\tt character*12$ name of the LCM object (type $\tt L_FLUX)$ containing the unperturbed flux used to decontaminate the GPT solution.
SYST	$\tt character*12$ name of the LCM object (type <code>L_SYSTEM</code>) containing the unperturbed system matrices.
TRACK	character*12 name of the LCM object (type L_TRACK) containing the TRACKING.

(gptflu_data) structure containing the data to module GPTFLU:.

1.11.1 Data input for module GPTFLU:

Table 24: Structure (gptflu_data)

```
[ EDIT iprint ]
[ { VAR1 | ACCE } icl1 icl2 ]
[ EXTE [ maxout ] [ epsout ] ]
[ THER [ maxthr ] [ epsthr ] ]
[ ADI nadi ]
[ { EXPLICIT | IMPLICIT } ]
FROM-TO { ALL | i<sub>src1</sub> i<sub>src2</sub> }
;
```

where	
EDIT	keyword used to set <i>iprint</i> .
iprint	index used to control the printing in module GPTFLU:. =0 for no print; =1 for minimum printing (default value); =2 iteration history is printed; =3 the solution is printed; =4 at each iteration, the new solution is compared to a reference solution previously stored in $FLUX_GPT$ under the name REF; =5 the convergence histogram is stored in $FLUX_GPT$.
VAR1	keyword used to set the parameters $(icl1 \text{ and } icl2)$ of the variational acceleration technique.
ACCE	alias keyword for VAR1.
icl1	number of free outer iterations in a cycle of the SVAT. The default value is $icl 1 = 3$.
icl2	number of accelerated outer iterations in a cycle of the SVAT. The default value is $icl2 = 3$. A convergence in free iterations is obtained by setting $icl1 = 200$ (or $icl1 = maxx0$) and $icl2 = 0$.
EXTE	keyword to specify that the control parameters for the external iteration are to be modified.
maxout	maximum number of external iterations. The fixed default value is $maxout = 200$.
epsout	convergence criterion for the external iterations. The fixed default value is $epsout = 1.0 \times 10^{-4}$. The outer iterations are stopped when the following criteria is reached:
	$\max_i \Gamma_i^{(k-1)} - \Gamma_i^{(k)} \le epsout \times \max_i \Gamma_i^{(k)} $
	where $\vec{\Gamma}^{(k)} = \operatorname{col}\{\Gamma_i^{(k)}; i = 1, I\}$ is the product of the \mathbb{B} matrix times the unknown vector at the k-th outer iteration.
THER	keyword to specify that the control parameters for the thermal iterations are to be modified.
maxthr	maximum number of thermal iterations. The fixed default value is $maxthr = 0$ corresponding to no thermal iterations.
epsthr	convergence criterion for the thermal iterations. The fixed default value is $epsthr = 1.0 \times 10^{-2}$.
ADI	keyword used to set <i>nadi</i> in cases where Trivac is used.
nadi	number of alternating direction implicit (ADI) inner iterations per outer iteration. The default value is $nadi = 1$. If this value causes a failure of the acceleration process, it is recommended that a larger value be tried. The optimal choice is generally the minimum value of $nadi$ which allows a convergence in less than 75 outer iterations. $nadi = 1$ or $nadi = 2$ is generally the best choice for production-type calculations. The greater $nadi$ is, the smaller the asymptotic convergence constant (ACC) becomes. Taking an arbitrary large value (e.g., $nadi = 20$) leads to numerical results identical to those obtained by inverting the system matrices at each outer iteration (at a prohibitive CPU cost). In this case, the ACC is almost equal to the dominance ratio of the iterative matrix.
EXPLICIT	keyword used to obtain the solution of an direct fixed source eigenvalue problem.
IMPLICIT	keyword used to obtain the solution of an adjoint fixed source eigenvalue problem. If neither 'EXPLICIT' nor 'IMPLICIT' are provided the default value will be chosen as a function of n_{var} and $n_{cst} + 1$.
FROM-TO	keyword used to specify the numbers of the sources for which a generalized adjoint will be calculated.
ALL	keyword used to recover all sources available in <i>GPT</i> .
i_{src1}	number of the first source.
i_{src1}	number of the last source.

1.12 The OUT: module

The OUT: module is used to compute the reaction rates and to store them in an extended MACROLIB (type L_MACROLIB) corresponding to a solution (type L_FLUX) of the matrix system. The calling specifications are:

Table 25: Structure (OUT:)

MACRO2 := OUT: FLUX TRACK MACRO GEOM :: (out_data)

where

MACRO2	$\texttt{character*12} \text{ name of the LCM object (type L_MACROLIB) containing the extended MACROLIB.}$
FLUX	$\tt character*12$ name of the LCM object (type <code>L_FLUX</code>) containing a solution.
TRACK	$character*12$ name of the LCM object (type L_TRACK) containing a TRACKING.
MACRO	$\texttt{character*12} \text{ name of the LCM object (type L_MACROLIB) containing the reference MACROLIB.}$
GEOM	$\texttt{character*12} \text{ name of the LCM object (type L_GEOM) containing the reference GEOMETRY.}$
(out_data)	structure containing the data to module OUT:.

1.12.1 Data input for module OUT:

Table 26: Structure (out_data)

```
[ EDIT iprint ]
[ MODE imode ]
[ { DIRE | ADJO } ]
[ POWR power ]
[ INTG { IN | MIX | (ihom(i), i=1,nreg )} ]
;
```

where	
EDIT	keyword used to set <i>iprint</i> .
iprint	index used to control the printing in module $\texttt{OUT:.}=0$ for no print; =1 for minimum printing (default value).
MODE	keyword to specify the flux harmonic index <i>imode</i> .
imode	index of the flux harmonic recovered by the OUT: module if the MONI keyword was set in module FLUD: (see Sect. 1.9.1). By default, it is assumed that the MONI keyword was not used.
DIRE	use the direct flux to perform homogenization (default value).

34

ADJO	use the adjoint flux to perform homogenization.
POWR	keyword used to set <i>power</i> .
power	value of the power in MW used to normalize the flux. By default, the flux is not normalized.
INTG	keyword used to compute the reaction rates.
IN	keyword for computing the reaction rates on the geometry mesh (see Sect. $1.3.1)$ before mesh-splitting.
MIX	keyword for computing the reaction rates on the mixture mesh previously used to define the geometry (see Sect. $1.3.1$) before mesh-splitting.
ihom	index of the homogenized region corresponding to the each region of the geometry (see Sect. $1.3.1)$ before mesh-splitting.

1.13 The ERROR: module

The ERROR: module is used to compare reaction rates contained into two extended MACROLIBS and to print statistics regarding the comparison.

The QUANDRY-type power densities are first compared. These power densities are defined by the following relation:

$$P_i^{\text{quandry}} = \frac{\sum_i V_i}{V_i} \frac{P_i}{\sum_i P_i}$$

where P_i is the total power and V_i is the volume of the region *i*. The maximum and averaged errors are respectively defined by:

$$\epsilon_{\max} = \max_{i} \frac{|P_i^{\text{quandry}} - P_i^{\text{quandry}*}|}{P_i^{\text{quandry}*}}$$

and

$$\bar{\epsilon} = \frac{1}{V_{\rm core}} \sum_{i} \left[\frac{|P_i^{\rm quandry} - P_i^{\rm quandry*}|}{P_i^{\rm quandry*}} \right] V_i$$

where $P_i^{\text{quandry}*}$ is computed using the reference powers (stored in *MACRO1*) and V_{core} is the total volume of the regions where the power density is not equal to zero.

The normalized removal rates $T_{i,g}^{\text{norm}}$ in each region i and energy group g are next computed using the following formula:

$$T_{i,g} = (\Sigma_{i,g} - \Sigma_{wi,g}) \ \phi_{i,g} V_i$$
$$T_{i,g}^{\text{norm}} = \frac{1}{\sum_i \sum_g T_{i,g}} \ T_{i,g}$$

where $\Sigma_{i,g}$ is the total macroscopic cross section, $\Sigma_{wi,g}$ is the within-group scattering cross section and $\phi_{i,g}$ is the neutron flux. The maximum and averaged errors are respectively defined by:

$$\epsilon_{\max g} = \max_{i} \frac{|T_{i,g}^{\text{norm}} - T_{i,g}^{\text{norm}*}|}{T_{i,q}^{\text{norm}*}}$$

and

$$\bar{\epsilon}_g = \frac{1}{N} \sum_i \left[\frac{|T_{i,g}^{\text{norm}} - T_{i,g}^{\text{norm}*}|}{T_{i,g}^{\text{norm}*}} \right]$$

where $T_{i,g}^{\text{norm}*}$ is computed using the reference values (stored in *MACRO1*) and N is the total number of regions in the MACROLIB.

The calling specifications are:

Table 27: Structure (ERROR:)

ERROR: MACRO1 MACRO2 :: [HREA hname] [NREG nreg] ;

IGE-293

MACRO1	$\label{eq:character*12} character*12 \ name of the \ \mbox{LCM} \ object \ (type \ \mbox{L-MACROLIB}) \ containing the extended \ \mbox{MACROLIB} used to \ compute the reference reaction rates.$
MACRO2	$\label{eq:character*12} have of the \mbox{LCM} object \mbox{(type L-MACROLIB)} containing the extended \mbox{MACROLIB} used to compute the approximate reaction rates.$
HREA	keyword used to set the character name <i>hname</i> .
hname	character*8 name of the nuclear reaction used to compute the power map. By default, reaction $\tt H-FACTOR$ is used.
NREG	keyword used to set the <i>nreg</i> number.

nreg integer number set to the number of regions used in statistics. By default, all available regions are used.

1.14 The INIKIN: module

The INIKIN: module is used to recover the steady-state solution and to initialize the kinetics parameters. The delayed neutron information can be provided directly from the input file or recovered from the MACROLIB data structure.

The initial presursor concentrations are obtained as a function of the strady-state solution. If $\phi_g(\mathbf{r}, t_0)$ is the initial flux in energy group g divided by k_{eff} , the corresponding initial conditions of the precursors are obtained as

$$c_{\ell}(\boldsymbol{r}, t_0) = \frac{1}{\lambda_{\ell}} \sum_{h=1}^{G} \nu \Sigma_{\mathrm{f}\ell, h}^{\mathrm{del}}(\boldsymbol{r}) \phi_h(\boldsymbol{r}, t_0); \quad \ell = 1, N_d.$$

$$(1.5)$$

where $\nu \Sigma_{\mathrm{f}\ell,h}^{\mathrm{del}}(\mathbf{r})$ is ν times the delayed macroscopic fission cross section in energy group h for precursor group ℓ .

The calling specifications are:

Table 28: Structure (INIKIN:)

KINET := INIKIN: MACRO TRACK SYST FLUX :: (inikin_data)

where

KINET	$\texttt{character*12}$ name of the LCM object (type <code>L_KINET</code>) to be created by the module.
MACRO	$\texttt{character*12} \text{ name of the LCM object (type L_MACROLIB) containing the MACROLIB information.}$
TRACK	$\texttt{character*12}$ name of the LCM object (type <code>L_TRACK</code>) containing the <code>TRACKING</code> information.
SYST	$\texttt{character*12} \text{ name of the LCM object (type L_SYSTEM) corresponding to MACROLIB MACRO and TRACKING TRACK.}$
FLUX	$\tt character*12$ name of the LCM object (type $\tt L_FLUX)$ containing the initial steady-state solution.

(inikin_data) structure containing the data to module INIKIN: (see Sect. 1.14.1).

1.14.1 Data input for module INIKIN:

Table 29: Structure (inikin_data)

```
[EDIT iprint]
[NGRP ngrp]
NDEL ndg
[BETA (beta(i), i=1,ndg)]
[LAMBDA (lambda(i), i=1,ndg)]
[CHID ((chid(i), i=1,ndg), j=1,ngrp)]
```

;

Structure (inikin_data)

```
[ NORM { fnorm | MAX | POWER-INI power } ]
```

where	
EDIT	keyword used to set <i>iprint</i> index.
iprint	integer index used to control the printing in module $\texttt{INIKIN}:$ =0 for no print; =1 for minimum printing (default value); larger values of <i>iprint</i> will produce increasing amounts of output.
NGRP	keyword used to set the $ngrp$ number. By default, this information is recovered from the solution object $FLUX.$
ngrp	integer total number of energy groups.
NDEL	keyword used to set the ndg number.
ndg	integer total number of the delayed neutron groups.
BETA	keyword used to indicate the reading of <i>beta</i> values from the input file. If these values are not provided, they should be recorded in the MACROLIB data structure.
beta	real array containing the delayed neutron fractions for each delayed group.
LAMBDA	keyword used to indicate the reading of <i>lambda</i> values from the input file. If these values are not provided, they should be recorded in the MACROLIB data structure.
lambda	real array containing the precursors decay constants for each delayed group.
CHID	keyword used to indicate the reading of <i>chid</i> values from the input file. If these values are not provided, they should be recorded in the MACROLIB data structure.
chid	real array representing the delayed multigroup fission spectrum.
NORM	keyword used to normalize the initial flux. By default, the flux is not normalized.
fnorm	real normalization factor.
MAX	keyword used to set the flux normalization factor to $1/f_{\rm max}$ where $f_{\rm max}$ is the maximum flux in the core.
POWER-INI	keyword used to set the flux normalization factor to a given value of the initial power.
power	real initial power in MW.

1.15 The KINSOL: module

The KINSOL: module is used to solve the space-time neutron kinetics equations at current time step of transient. Several implicit numerical schemes are available for this purpose. Consider first the differential equation for precursor concentrations:

$$\frac{\partial c_{\ell}(\boldsymbol{r},t)}{\partial t} + \lambda_{\ell} c_{\ell}(\boldsymbol{r},t) = \sum_{h=1}^{G} \nu \Sigma_{\mathrm{f}\ell,h}^{\mathrm{del}}(\boldsymbol{r}) \phi_{h}(\boldsymbol{r},t); \quad \ell = 1, N_{d}.$$
(1.6)

Consider a solution between times t_{n-1} and $t_n = t_{n-1} + \Delta t_n$. First, an analytic solution can be obtained by assuming a ramp variation of the fission reaction rates over time step Δt_n . This solution is written

$$c_{\ell}(\boldsymbol{r},t_{n}) = c_{\ell}(\boldsymbol{r},t_{n-1}) e^{-\lambda_{\ell} \Delta t_{n}} + \frac{F_{\ell}(\boldsymbol{r},t_{n-1})}{\lambda_{\ell}} \left[\frac{1}{\lambda_{\ell} \Delta t_{n}} \left(1 - e^{-\lambda_{\ell} \Delta t_{n}} \right) - e^{-\lambda_{\ell} \Delta t_{n}} \right] + \frac{F_{\ell}(\boldsymbol{r},t_{n})}{\lambda_{\ell}} \left[1 - \frac{1}{\lambda_{\ell} \Delta t_{n}} \left(1 - e^{-\lambda_{\ell} \Delta t_{n}} \right) \right]$$

$$(1.7)$$

where the delayed fission reaction rates are defined as

$$F_{\ell}(\boldsymbol{r},t_n) = \sum_{h=1}^{G} \nu \Sigma_{\mathrm{f}\ell,h}^{\mathrm{del}}(\boldsymbol{r}) \phi_h(\boldsymbol{r},t_n) = \beta_{\ell} \sum_{h=1}^{G} \nu \Sigma_{\mathrm{f}h}(\boldsymbol{r}) \phi_h(\boldsymbol{r},t_n).$$
(1.8)

An implicit theta solution is presented in Chapter 5 of Ref. 1. This solution is written

$$c_{\ell}(\boldsymbol{r}, t_{n}) = \left[\frac{1 - (1 - \Theta_{p})\lambda_{\ell}\Delta t_{n}}{1 + \Theta_{p}\lambda_{\ell}\Delta t_{n}}\right]c_{\ell}(\boldsymbol{r}, t_{n-1}) + \frac{F_{\ell}(\boldsymbol{r}, t_{n-1})}{\lambda_{\ell}}\left[\frac{(1 - \Theta_{p})\lambda_{\ell}\Delta t_{n}}{1 + \Theta_{p}\lambda_{\ell}\Delta t_{n}}\right] + \frac{F_{\ell}(\boldsymbol{r}, t_{n})}{\lambda_{\ell}}\left[\frac{\Theta_{p}\lambda_{\ell}\Delta t_{n}}{1 + \Theta_{p}\lambda_{\ell}\Delta t_{n}}\right]$$
(1.9)

where $\Theta_{\rm p}$ is the theta-factor for precursors.

The fixed-source corresponding to the analytic solution for precursors is written

$$S_{g}^{\text{exact}}(\boldsymbol{r},t_{n}) = \frac{1}{V_{n,g}\Delta t_{n}} \phi_{g}(\boldsymbol{r},t_{n-1}) + \sum_{\ell} \lambda_{\ell} \left[1 - \Theta_{f} + \Theta_{f} e^{-\lambda_{\ell} \Delta t_{n}}\right] \chi_{\ell,g}^{\text{del}}(\boldsymbol{r}) c_{\ell}(\boldsymbol{r},t_{n-1}) + (1 - \Theta_{f}) \left\{ \boldsymbol{\nabla} \cdot \mathbb{D}_{g}(\boldsymbol{r}) \boldsymbol{\nabla} \phi_{g}(\boldsymbol{r},t_{n-1}) - \Sigma_{rg}(\boldsymbol{r}) \phi_{g}(\boldsymbol{r},t_{n-1}) \right. + \left. \sum_{\substack{h=1\\h \neq g}}^{G} \Sigma_{g \leftarrow h}(\boldsymbol{r}) \phi_{h}(\boldsymbol{r},t_{n-1}) + \chi_{g}^{\text{ss}}(\boldsymbol{r}) F(\boldsymbol{r},t_{n-1}) \right\} - \left. \sum_{\ell} \left[1 - \Theta_{f} - \Theta_{f} \left(\frac{1}{\lambda_{\ell} \Delta t_{n}} \left(1 - e^{-\lambda_{\ell} \Delta t_{n}} \right) - e^{-\lambda_{\ell} \Delta t_{n}} \right) \right] \chi_{\ell,g}^{\text{del}}(\boldsymbol{r}) F_{\ell}(\boldsymbol{r},t_{n-1})$$
(1.10)

where the steady-state fission reaction rates are defined as

$$F(\boldsymbol{r}, t_n) = \sum_{h=1}^{G} \nu \Sigma_{\mathrm{f}h}(\boldsymbol{r}) \,\phi_h(\boldsymbol{r}, t_n).$$
(1.11)

The fixed-source corresponding to the implicit theta solution is presented in Chapter 5 of Ref. 1 and is written

$$S_{g}^{\Theta}(\boldsymbol{r},t_{n}) = \frac{1}{V_{n,g}\Delta t_{n}}\phi_{g}(\boldsymbol{r},t_{n-1}) + \sum_{\ell}\lambda_{\ell}\left[1 - \Theta_{f} + \Theta_{f}\frac{1 - (1 - \Theta_{p})\lambda_{\ell}\Delta t_{n}}{1 + \Theta_{p}\lambda_{\ell}\Delta t_{n}}\right]\chi_{\ell,g}^{del}(\boldsymbol{r})c_{\ell}(\boldsymbol{r},t_{n-1}) + (1 - \Theta_{f})\left\{\boldsymbol{\nabla}\cdot\mathbb{D}_{g}(\boldsymbol{r})\boldsymbol{\nabla}\phi_{g}(\boldsymbol{r},t_{n-1}) - \Sigma_{rg}(\boldsymbol{r})\phi_{g}(\boldsymbol{r},t_{n-1})\right. + \sum_{\substack{h=1\\h\neq g}}^{G}\Sigma_{g\leftarrow h}(\boldsymbol{r})\phi_{h}(\boldsymbol{r},t_{n-1}) + \chi_{g}^{ss}(\boldsymbol{r})F(\boldsymbol{r},t_{n-1})\right\} - \sum_{\ell}\left[1 - \Theta_{f} - \Theta_{f}\frac{(1 - \Theta_{p})\lambda_{\ell}\Delta t_{n}}{1 + \Theta_{p}\lambda_{\ell}\Delta t_{n}}\right]\chi_{\ell,g}^{del}(\boldsymbol{r})F_{\ell}(\boldsymbol{r},t_{n-1}).$$
(1.12)

The flux equation at end-of-step is now presented. The equation corresponding to the analytic solution for precursors is written

$$\frac{1}{V_{n,g}\Delta t_{n}}\phi_{g}(\boldsymbol{r},t_{n}) - \Theta_{f}\boldsymbol{\nabla}\cdot\mathbb{D}_{g}(\boldsymbol{r})\boldsymbol{\nabla}\phi_{g}(\boldsymbol{r},t_{n}) + \Theta_{f}\Sigma_{rg}(\boldsymbol{r})\phi_{g}(\boldsymbol{r},t_{n})$$

$$= S_{g}^{exact}(\boldsymbol{r},t_{n}) + \Theta_{f}\sum_{\substack{h=1\\h\neq g}}^{G}\Sigma_{g\leftarrow h}(\boldsymbol{r})\phi_{h}(\boldsymbol{r},t_{n})$$

$$+ \Theta_{f}\chi_{g}^{ss}(\boldsymbol{r})F(\boldsymbol{r},t_{n}) - \Theta_{f}\sum_{\ell}\chi_{\ell,g}^{del}(\boldsymbol{r})\frac{1}{\lambda_{\ell}\Delta t_{n}}\left(1 - e^{-\lambda_{\ell}\Delta t_{n}}\right)F_{\ell}(\boldsymbol{r},t_{n}).$$
(1.13)

The equation corresponding to the implicit theta solution is presented in Chapter 5 of Ref. 1 and is written

$$\frac{1}{V_{n,g}\Delta t_n}\phi_g(\boldsymbol{r},t_n) - \Theta_{\rm f}\boldsymbol{\nabla}\cdot\mathbb{D}_g(\boldsymbol{r})\boldsymbol{\nabla}\phi_g(\boldsymbol{r},t_n) + \Theta_{\rm f}\Sigma_{\rm rg}(\boldsymbol{r})\phi_g(\boldsymbol{r},t_n) \\
= S_g^{\Theta}(\boldsymbol{r},t_n) + \Theta_{\rm f}\sum_{\substack{h=1\\h\neq g}}^G \Sigma_{g\leftarrow h}(\boldsymbol{r})\phi_h(\boldsymbol{r},t_n) \\
+ \Theta_{\rm f}\chi_g^{\rm ss}(\boldsymbol{r})F(\boldsymbol{r},t_n) - \Theta_{\rm f}\sum_{\ell}\chi_{\ell,g}^{\rm del}(\boldsymbol{r})\frac{1}{1+\Theta_{\rm p}\lambda_\ell\Delta t_n}F_\ell(\boldsymbol{r},t_n).$$
(1.14)

The calling specifications are:

Table 30: Structure (KINSOL:)

KINET := KINSOL: KINET MACRO TRACK SYST [MACRO_0 SYST_0] :: (kinsol_data)

where

<i>KINET</i> character*12 name of the LCM object (typ	ype L_KINET)) in modification mode.
---	--------------	-------------------------

- MACRO character*12 name of the LCM object (type L_MACROLIB) containing the MACROLIB information corresponding to the current time step of a transient.
- TRACK character*12 name of the LCM object (type L_TRACK) containing the TRACKING information.
- SYST character*12 name of the LCM object (type L_SYSTEM) corresponding to MACROLIB MACRO and TRACKING TRACK.
- MACRO_0 character*12 name of the LCM object (type L_MACROLIB) containing the MACROLIB information corresponding to the beginning-of-step conditions in case a ramp variation of the cross sections in set. Beginning-of-step conditions should not be confused with beginningof-transient or initial conditions. By default, a step variation is set where cross sections are assumed constant and given by MACRO.
- SYST_0 character*12 name of the LCM object (type L_SYSTEM) corresponding to MACROLIB MACRO_0 and TRACKING TRACK.

(kinsol_data) structure containing the data to module KINSOL: (see Sect. 1.15.1).

1.15.1 Data input for module KINSOL:

Table 31: Structure (kinsol_data)

```
[ EDIT iprint ]
DELTA delta
SCHEME FLUX { IMPLIC | CRANK | THETA ttflx }
PREC { IMPLIC | CRANK | EXPON | THETA ttprc }
[ { VAR1 | ACCE } icl1 icl2 ]
[ EXTE [ maxout ] [ epsout ] ]
[ THER [ maxthr ] [ epsthr ] ]
[ ADI nadi ]
;
```

where

EDIT	keyword used to set <i>iprint</i> index.
iprint	integer index used to control the printing in module $KINSOL:$ =0 for no print; =1 for minimum printing (default value); larger values of <i>iprint</i> will produce increasing amounts of output.
DELTA	keyword used to set the <i>delta</i> value.
delta	current time increment Δt_n of transient.
SCHEME	keyword used to indicate the temporal numerical schemes.
FLUX	keyword used to select the temporal scheme for the fluxes equations.
PREC	keyword used to select the temporal scheme for the precursors equations.
IMPLIC	keyword used to indicate the full implicit temporal scheme.
CRANK	keyword used to indicate the Crank-Nicholson temporal scheme.
EXPON	keyword used to indicate the analytical integration scheme for precursors equations.
THETA	keyword used to indicate the general temporal scheme according to the <i>theta</i> method.
ttflx	value of theta parameter $\Theta_{\rm f}$ for the flux equations. This value should be greater than 0.5 and less than 1.0.
ttprc	value of theta parameter Θ_p for the precursors equations. This value should be greater than 0.5 and less than 1.0.
VAR1	keyword used to set the parameters $(icl1 \text{ and } icl2)$ of the symmetrical variational acceleration technique (SVAT).
ACCE	alias keyword for VAR1.
icl1	number of free outer iterations in a cycle of the SVAT. The default value is $icl 1 = 3$.
icl2	number of accelerated outer iterations in a cycle of the SVAT. The default value is $icl 2 = 3$. A convergence in free iterations is obtained by setting $icl 1 = 200$ (or $icl 1 = maxx0$) and $icl 2 = 0$.

EXTE	keyword t	o specify	that the co	ntrol parar	neters for t	the external	iteration a	are to	be modified.
	•/	•/							

maxout maximum number of external iterations. The fixed default value is maxout = 200.

epsout convergence criterion for the external iterations. The fixed default value is $epsout = 1.0 \times 10^{-4}$. The outer iterations are stopped when the following criteria is reached:

$$\max_{i} |\Phi_i^{(k-1)} - \Phi_i^{(k)}| \leq epsout \times \max_{i} |\Phi_i^{(k)}|$$

where $\vec{\Phi}^{(k)} = \operatorname{col}\{\Phi_i^{(k)}; i = 1, I\}$ is the product of the *B* matrix times the unknown vector at the *k*-th outer iteration.

- THER keyword to specify that the control parameters for the thermal iterations are to be modified.
- maxthr maximum number of thermal iterations. The fixed default value is maxthr = 0 corresponding to no thermal iterations.
- epsthr convergence criterion for the thermal iterations. The fixed default value is $epsthr = 1.0 \times 10^{-2}$.
- ADI keyword used to set *nadi* in cases where Trivac is used.
- nadi number of alternating direction implicit (ADI) inner iterations per outer iteration. The default value is nadi = 1. If this value causes a failure of the acceleration process, it is recommended that a larger value be tried. The optimal choice is generally the minimum value of nadi which allows a convergence in less than 75 outer iterations. nadi = 1 or nadi = 2 is generally the best choice for production-type calculations. The greater nadi is, the smaller the asymptotic convergence constant (ACC) becomes. Taking an arbitrary large value (e.g., nadi = 20) leads to numerical results identical to those obtained by inverting the system matrices at each outer iteration (at a prohibitive CPU cost). In this case, the ACC is almost equal to the dominance ratio of the iterative matrix. The default value is recovered in the state vector of the TRACKING object TRACK.

1.16 The VAL: module

The VAL: module supplies an interpolation of the flux in diffusion calculations for Cartesian geometries. The calling specifications are:

Table 32: Structure (VAL:)

IFLU := VAL: TRKNAM FLUNAM :: (descval)

where

IFLU	$\verb character*12 name of the INTERPFLUX data structure (\texttt{L-FVIEW} signature) where the interpolated flux distribution will be stored.$
TRKNAM	<code>character*12</code> name of the read-only <code>TRACKING</code> data structure (<code>L_TRACK</code> signature) containing the tracking.
FLUNAM	$\tt character*12$ name of the read-only <code>FLUXUNK</code> data structure (L_FLUX signature) containing a transport solution.
(descval)	structure containing the input data to this module to compute interpolated flux (see Section $1.16.1$).

1.16.1 Data input for module VAL:

Table 33: Structure (descval)

 $\begin{array}{l} [\text{ EDIT } iprint \] \\ [\text{ MODE } imode \] \\ \text{DIM } dim \ (dxyz(i), \ i=1, dim) \\ ; \end{array}$

where

EDIT	keyword used to modify the print level <i>iprint</i> .
iprint	integer index used to control the printing in module VAL:. =0 for no print; =1 for minimum printing (default value); larger values of <i>iprint</i> will produce increasing amounts of output.
MODE	keyword to specify the flux harmonic index <i>imode</i> .
imode	index of the flux harmonic recovered by the VAL: module if the MONI keyword was set in module FLUD: (see Sect. 1.9.1). By default, it is assumed that the MONI keyword was not used.
DIM	keyword to specify the number <i>dim</i> .
dim	number of dimension of the geometry.

dxyz mesh interval along each direction which is used to define the grid where the flux is interpolated.

2 EXAMPLES OF INPUT DATA FILES

2.1 IAEA-2D benchmark

The IAEA-2D benchmark is defined in Refs. 3, 18 and its geometry is represented in Fig. 11. Here, it is solved using a parabolic variational collocation method without mesh splitting of the elements:



Figure 11: Description of the IAEA-2D benchmark.

```
LINKED_LIST IAEA MACRO TRACK SYSTEM FLUX EDIT ;
MODULE GEO: MAC: TRIVAT: TRIVAA: FLUD: OUT: END: ;
IAEA := GEO: :: CAR2D 9 9
          EDIT 2
          X- DIAG X+ VOID
          Y- SYME Y+ DIAG
          MIX 322232214
                 2 2 2 2 2 2 1 4
                   2 2 2 2 1 1 4
                     2 2 2 1 4 4
                       3 1 1 4 0
                          1 4 4 0
                           4 0 0
                             0 0
                               0
          MESHX 0.0 20.0 40.0 60.0 80.0 100.0 120.0 140.0 160.0 180.0
           ;
MACRO := MAC: ::
EDIT 2 NGRO 2 NMIX 4
READ INPUT
MIX
        1
     DIFF 1.500E+00 4.0000E-01
    TOTAL 3.012E-02 8.0032E-02
   NUSIGF
           0.000E+00 1.3500E-01
 H-FACTOR 0.000E+00 1.3500E-01
      SCAT
           1 1 0.0 2 2 0.0 0.2E-01
MIX
        2
     DIFF 1.500E+00 4.0000E-01
```

```
TOTAL 3.012E-02 8.5032E-02
   NUSIGF 0.000E+00 1.3500E-01
 H-FACTOR 0.000E+00 1.3500E-01
     SCAT 1 1 0.0 2 2 0.0 0.2E-01
MIX
        3
     DIFF 1.500E+00 4.00000E-01
    TOTAL 3.012E-02 1.30032E-01
   NUSIGF 0.000E+00 1.35000E-01
 H-FACTOR 0.000E+00 1.35000E-01
     SCAT 1 1 0.0 2 2 0.0 0.2E-01
MIX
        4
     DIFF 2.000E+00 3.0000E-01
    TOTAL 4.016E-02 1.0024E-02
     SCAT 1 1 0.0 2 2 0.0 0.4E-01
 ;
TRACK := TRIVAT: IAEA ::
     TITLE 'IAEA-2D BENCHMARK'
     MAXR 81 PRIM 2 ;
SYSTEM := TRIVAA: MACRO TRACK :: ;
FLUX := FLUD: SYSTEM ::
     EDIT 2 ;
EDIT := OUT: FLUX ::
      EDIT 2 INTG
      1 2 3 4 5 6 7 8
                             0
         9 10 11 12 13 14 15 0
           16 17 18 19 20 21 0
              22 23 24 25 0
                             0
                 26 27 28 0
                             0
                    29 0 0 0
                          0 0
                        0
                           0
                             0
                              0
      ;
```

```
END: ;
```

2.2 Biblis-2D benchmark

The rods-withdrawn configuration of the Biblis-2D benchmark is defined in Ref. 3 and its geometry is represented in Fig. 12. Here, it is solved using a parabolic variational collocation method without mesh splitting of the elements:

```
LINKED_LIST BIBLIS MACRO TRACK SYSTEM FLUX EDIT ;

MODULE GEO: MAC: TRIVAT: TRIVAA: FLUD: OUT: END: ;

*

BIBLIS := GEO: :: CAR2D 9 9

EDIT 2

X- DIAG X+ VOID

Y- SYME Y+ DIAG

MIX 1 8 2 6 1 7 1 4 3

1 8 2 8 1 1 4 3

1 8 2 7 1 4 3

2 8 1 8 4 3

2 5 4 3 3

4 4 3 0

3 3 0

0 0
```



Figure 12: Description of the Biblis-2D benchmark, rods-withdrawn configuration.

```
0
      MESHX 0.0 23.1226 46.2452 69.3678 92.4904 115.613 138.7356
            161.8582 184.9808 208.1034
      ;
MACRO := MAC: ::
EDIT 2 NGRO 2 NMIX 8
READ INPUT
MIX
        1
     DIFF
           1.436000E+00 3.635000E-01
    TOTAL
          2.725820E-02 7.505800E-02
   NUSIGF 5.870800E-03 9.606700E-02
 H-FACTOR
           2.376800E-03
                         3.889400E-02
           1 1 0.0 2 2 0.0 1.775400E-02
     SCAT
MIX
        2
     DIFF
           1.436600E+00 3.636000E-01
    TOTAL
           2.729950E-02 7.843600E-02
   NUSIGF
           6.190800E-03 1.035800E-01
 H-FACTOR
           2.506400E-03 4.193500E-02
     SCAT
           1 1 0.0 2 2 0.0 1.762100E-02
MIX
        3
     DIFF
           1.320000E+00 2.772000E-01
    TOTAL
           2.576220E-02 7.159600E-02
     SCAT
          1 1 0.0 2 2 0.0 2.310600E-02
MIX
        4
     DIFF
           1.438900E+00 3.638000E-01
    TOTAL
           2.746400E-02 9.140800E-02
   NUSIGF
           7.452700E-03
                         1.323600E-01
 H-FACTOR 3.017300E-03 5.358700E-02
     SCAT
           1 1 0.0 2 2 0.0 1.710100E-02
MIX
        5
     DIFF
           1.438100E+00 3.665000E-01
    TOTAL
          2.729300E-02 8.482800E-02
   NUSIGF
           6.190800E-03 1.035800E-01
 H-FACTOR 2.506400E-03 4.193500E-02
     SCAT
           1 1 0.0 2 2 0.0 1.729000E-02
MIX
        6
```

```
DIFF 1.438500E+00 3.665000E-01
    TOTAL 2.732400E-02 8.731400E-02
   NUSIGF 6.428500E-03 1.091100E-01
 H-FACTOR 2.602600E-03 4.417400E-02
    SCAT 1 1 0.0 2 2 0.0 1.719200E-02
MIX 7
    DIFF 1.438900E+00 3.679000E-01
    TOTAL 2.729000E-02 8.802400E-02
   NUSIGF 6.190800E-03 1.035800E-01
 H-FACTOR 2.506400E-03 4.193500E-02
    SCAT 1 1 0.0 2 2 0.0 1.712500E-02
MIX 8
    DIFF 1.439300E+00 3.680000E-01
    TOTAL 2.732100E-02 9.051000E-02
   NUSIGF 6.428500E-03 1.091100E-01
 H-FACTOR 2.602600E-03 4.417400E-02
     SCAT 1 1 0.0 2 2 0.0 1.702700E-02
     ;
TRACK := TRIVAT: BIBLIS ::
     TITLE 'BIBLIS BENCHMARK'
     EDIT 5 MAXR 81 PRIM 2 ;
SYSTEM := TRIVAA: MACRO TRACK ::
     EDIT 5 ;
FLUX := FLUD: SYSTEM ::
     EDIT 2 ;
EDIT := OUT: FLUX ::
      EDIT 2 INTG
      1 2 3 4 5 6 7 8 0
         9 10 11 12 13 14 15 0
           16 17 18 19 20 21 0
              22 23 24 25 26 0
                27 28 29 0 0
                   30 31 0 0
                       0 0 0
                         0 0
                            0
      ;
END: ;
```

2.3 IAEA-3D benchmark

The IAEA-3D benchmark is defined in Ref. 18 and its geometry is represented in Fig. 13. Here, it is solved using a cubic mixed-dual method with mesh splitting of the second axial plane:



Figure 13: Description of the IAEA-3D benchmark.

```
LINKED_LIST IAEA3D MACRO TRACK SYSTEM FLUX EDIT ;
MODULE GEO: MAC: TRIVAT: TRIVAA: FLUD: OUT: END: ;
IAEA3D := GEO: :: CAR3D 9 9 4
         EDIT 2
         X- DIAG
                 X+ VOID
         Y- SYME
                 Y+ DIAG
         Z- VOID Z+ VOID
         MESHX 0.0 20.0 40.0 60.0 80.0 100.0 120.0 140.0 160.0 180.0
         MESHZ 0.0 20.0 280.0 360.0 380.0
         SPLITZ 1 2 1 1
         (* PLANE NB 1 *)
         MIX 4 4 4 4 4 4 4 4 4
               4 4 4 4 4 4 4 4
                 444444
                   44444
                     4 4 4 4 0
                       4440
                         4 0 0
```

```
DIFF 2.000E+00 3.0000E-01
    TOTAL 4.000E-02 1.0000E-02
    SCAT 1 1 0.0 2 2 0.0 0.4E-01
MIX 5
    DIFF 2.000E+00 3.0000E-01
    TOTAL 4.000E-02 5.5000E-02
    SCAT 1 1 0.0 2 2 0.0 0.4E-01
;
TRACK := TRIVAT: IAEA3D ::
     TITLE 'TEST IAEA 3D'
     EDIT 5 MAXR 405 DUAL 3 1 ;
SYSTEM := TRIVAA: MACRO TRACK ::
     EDIT 5 ;
FLUX := FLUD: SYSTEM ::
     EDIT 2 ;
EDIT := OUT: FLUX ::
     EDIT 2 INTG
      (* PLANE NB 1 *)
      0 0 0 0 0 0 0 0 0
        0 0 0 0 0 0 0 0
           0 0 0 0 0 0 0
              0 0 0 0 0 0
                0 0 0 0 0
                   0 0 0 0
                      0 0 0
                        0 0
                           0
      (* PLANE NB 2 *)
      1 2 3 4 5 6 7 8 0
        9 10 11 12 13 14 15 0
          16 17 18 19 20 21 0
             22 23 24 25 0 0
                26 27 28 0 0
                  29 0 0 0
                      0 0 0
                        0 0
                           0
      (* PLANE NB 3 *)
      30 31 32 33 34 35 36 37 0
        38 39 40 41 42 43 44 0
           45 46 47 48 49 50 0
              51 52 53 54 0 0
                55 56 57 0 0
                   58 0 0 0
                       0 0 0
                         0 0
                           0
      (* PLANE NB 4 *)
      0 0 0 0 0 0 0 0 0
        0 0 0 0 0 0 0 0
           0 0 0 0 0 0 0
              0 0 0 0 0 0
                0 0 0 0 0
                   0 0 0 0
                      0 0 0
```

```
0 0
0
;
END: ;
```

2.4 S30 hexagonal benchmark in 2-D

The S30 hexagonal benchmark in 2-D is defined in Ref. 14. Its geometry is represented in Fig. 14. Here, it is solved using a mesh centered finite difference method without mesh splitting of the hexagonal elements:



Figure 14: Description of the S30 hexagonal benchmark.

```
LINKED_LIST HEX MACRO TRACK SYSTEM FLUX EDIT ;
MODULE GEO: MAC: TRIVAT: TRIVAA: FLUD: OUT: END: ;
HEX := GEO: :: HEX
                     6
       EDIT 2
       HBC S30 ZERO
       SIDE 13.044
       MIX
       1
       2
       2
         2
       3
         3
       ;
MACRO := MAC: ::
 EDIT 2 NGRO 2 NMIX 3
 READ INPUT
 MIX
         1
      DIFF
           1.5E+00
                    4.00E-01
                    1.30E-01
     TOTAL
           3.0E-02
    NUSIGF
           0.0E+00
                    1.35E-01
  H-FACTOR
           0.0E+00 1.35E-01
      SCAT
            1 1 0.0 2 2 0.0 0.2E-01
 MIX
         2
      DIFF
            1.5E+00
                    4.00E-01
           3.0E-02 8.50E-02
     TOTAL
    NUSIGF
           0.0E+00
                    1.35E-01
  H-FACTOR
           0.0E+00 1.35E-01
            1 1 0.0 2 2 0.0 0.2E-01
      SCAT
 MIX
         3
      DIFF
           2.0E+00 3.0E-01
     TOTAL 4.0E-02 1.0E-02
      SCAT 1 1 0.0 2 2 0.0 0.4E-01
```

```
;

TRACK := TRIVAT: HEX ::

TITLE 'S30 HEXAGONAL BENCHMARK IN 2-D.'

EDIT 5 MAXR 50 MCFD (* IELEM= *) 1 (* ISPLH= *) 1 ;

SYSTEM := TRIVAA: MACRO TRACK ::

EDIT 5 ;

FLUX := FLUD: SYSTEM ::

EDIT 2 ;

EDIT 2 ;

EDIT := OUT: FLUX ::

EDIT 2 INTG IN ;

END: ;
```

2.5 LMW benchmark in 2-D

The LMW benchmark in 2-D is a space-time kinetics problem introduced by Greenman^[19] and used by Monier^[13]. Its geometry is represented in Fig. 15. Here, it is solved using a parabolic nodal collocation method with 2×2 mesh splitting of each element. A reactivity transient is induced by the rapid withdrawal of the control rod in material mixture 6. The control rod is removed in 26.7 s, causing a negative ramp variation in total cross section.



Figure 15: Description of the LMW benchmark in 2-D.

```
*----
* TEST CASE LMW 2D
*
* REF: G. Greenman, "A Quasi-Static Flux Synthesis Temporal Integration
* Scheme for an Analytic Nodal Method," Nuclear Engineer's Thesis,
* Massachusetts Institute of Technology, Department of Nuclear
* Engineering (May 1980).
*
*----
* Define STRUCTURES and MODULES used
*----
LINKED_LIST LMW TRACK MACRO1 SYSTEM1 MACRO2 SYSTEM2 FLUX KINET ;
MODULE GEO: MAC: TRIVAT: TRIVAA: FLUD: INIKIN: KINSOL: GREP: DELETE:
END: ;
```

```
IGE-293
```

```
REAL fnorm sigt1 sigt2 ;
REAL TIME := 0.0;
PROCEDURE assertS assertS2 ;
LMW := GEO: :: CAR2D 6 6
     X- REFL X+ ZERO
     Y- REFL Y+ ZERO
     MIX 1 1 1 2 3 4
         1 1 1 1 3 4
         1 1 5 1 3 4
         6 1 1 3 3 4
         3 3 3 3 4 4
         4 4 4 4 4 0
     MESHX 0.0 10. 30. 50. 70. 90. 110.
     MESHY 0.0 10. 30. 50. 70. 90. 110.
     SPLITX 2 2 2 2 2 2 2
     SPLITY 2 2 2 2 2 2 2
     ;
MACRO1 := MAC: ::
EDIT O NGRO 2 NMIX 6
READ INPUT
MIX
      1
     DIFF 1.423910E+00 3.563060E-01
    TOTAL 2.795756E-02 8.766216E-02
   NUSIGF 6.477691E-03 1.127328E-01
 H-FACTOR 2.591070E-03 4.509310E-02
     SCAT 1 1 0.0 2 2 0.0 0.175555E-01
     OVERV 0.800E-07 4.000E-06
MIX
       2
     DIFF 1.423910E+00 3.563060E-01
    TOTAL 2.850756E-02 9.146219E-02
   NUSIGF 6.477691E-03 1.127328E-01
  H-FACTOR 2.591070E-03 4.509310E-02
     SCAT 1 1 0.0 2 2 0.0 0.175555E-01
     OVERV 0.800E-07 4.000E-06
MIX
       3
     DIFF 1.425610E+00 3.505740E-01
    TOTAL 2.817031E-02 9.925634E-02
   NUSIGF 7.503282E-03 1.378004E-01
 H-FACTOR 3.001310E-03 5.512106E-02
     SCAT 1 1 0.0 2 2 0.0 0.171777E-01
     OVERV 0.800E-07 4.000E-06
MIX
      4
     DIFF 1.634220E+00 2.640020E-01
    TOTAL 3.025750E-02 4.936351E-02
     SCAT 1 1 0.0 2 2 0.0 0.275969E-01
     OVERV 0.800E-07 4.000E-06
MIX
     5
     DIFF 1.423910E+00 3.563060E-01
    TOTAL 2.795756E-02 8.766216E-02
   NUSIGF 6.477691E-03 1.127328E-01
 H-FACTOR 2.591070E-03 4.509310E-02
     SCAT 1 1 0.0 2 2 0.0 0.175555E-01
     OVERV 0.800E-07 4.000E-06
MIX 6
```

```
DIFF 1.423910E+00 3.563060E-01
     TOTAL 2.850756E-02 9.146217E-02
    NUSIGF 6.477691E-03 1.127328E-01
  H-FACTOR 2.591070E-03 4.509310E-02
      SCAT 1 1 0.0 2 2 0.0 0.175555E-01
      OVERV 0.800E-07 4.000E-06
TRACK := TRIVAT: LMW ::
      TITLE 'LMW 2-D BENCHMARK'
     EDIT 1 MAXR 144 MCFD 2 ;
SYSTEM1 := TRIVAA: MACRO1 TRACK ::
     EDIT 1 UNIT ;
FLUX := FLUD: SYSTEM1 TRACK ::
     EDIT 1 EXTE 5.0E-7 ;
assertS FLUX :: 'K-EFFECTIVE' 1 1.014803 ;
*----
* Crank-Nicholson space-time kinetics
*----
EVALUATE TIME := 0.0 ;
KINET := INIKIN: MACRO1 TRACK SYSTEM1 FLUX :: EDIT 1
     NDEL 6
      BETA 0.000247 0.0013845 0.001222 0.0026455 0.000832 0.000169
     LAMBDA 0.0127 0.0317 0.115 0.311 1.40 3.87
      CHID 1.0 1.0 1.0 1.0 1.0 1.0
            0.0 0.0 0.0 0.0 0.0 0.0
     NORM POWER-INI 1.0E4 ;
EVALUATE sigt1 := 2.850756E-02 ;
EVALUATE sigt2 := 9.146217E-02 ;
WHILE TIME 26.7 <= DO
  EVALUATE sigt1 := sigt1 5.5E-4 0.1 26.7 / * - ;
  EVALUATE sigt2 := sigt2 3.8E-3 0.1 26.7 / * - ;
  MACRO2 := MAC: MACRO1 ::
      EDIT O
      READ INPUT
      MIX 6
        TOTAL <<sigt1>> <<sigt2>>
      :
  SYSTEM2 := TRIVAA: MACRO2 TRACK ::
      EDIT 1 UNIT ;
  KINET := KINSOL: KINET MACRO2 TRACK SYSTEM2 MACRO1 SYSTEM1 ::
     EDIT 5 DELTA 0.1
      SCHEME FLUX CRANK PREC CRANK EXTE 1.0E-6 ;
  GREP: KINET :: GETVAL 'TOTAL-TIME' 1 >>TIME<< ;
  ECHO "TIME=" TIME "S" "sigt=" sigt1 sigt2 ;
  IF TIME 1.0 - ABS 1.0E-3 < THEN
    assertS2 KINET :: 'CTRL-FLUX' 1 1.986270E+02 ;
    assertS2 KINET :: 'CTRL-PREC' 1 1.095509E-01 ;
    assertS2 KINET :: 'E-POW' 1 1.008753E+04 ;
  ELSEIF TIME 5.0 - ABS 1.0E-3 < THEN
    assertS2 KINET :: 'CTRL-FLUX' 1 2.090369E+02 ;
    assertS2 KINET :: 'CTRL-PREC' 1 1.097266E-01 ;
    assertS2 KINET :: 'E-POW' 1 1.063990E+04 ;
  ELSEIF TIME 10.0 - ABS 1.0E-3 < THEN
    assertS2 KINET :: 'CTRL-FLUX' 1 2.305455E+02 ;
    assertS2 KINET :: 'CTRL-PREC' 1 1.104699E-01 ;
```

```
assertS2 KINET :: 'E-POW' 1 1.176902E+04 ;
 ELSEIF TIME 15.0 - ABS 1.0E-3 < THEN
   assertS2 KINET :: 'CTRL-FLUX' 1 2.641221E+02 ;
   assertS2 KINET :: 'CTRL-PREC' 1 1.121002E-01 ;
   assertS2 KINET :: 'E-POW' 1 1.352433E+04 ;
 ELSEIF TIME 20.0 - ABS 1.0E-3 < THEN
   assertS2 KINET :: 'CTRL-FLUX' 1 3.157370E+02 :
   assertS2 KINET :: 'CTRL-PREC' 1 1.150681E-01 ;
   assertS2 KINET :: 'E-POW' 1 1.621938E+04 ;
 ELSEIF TIME 25.0 - ABS 1.0E-3 < THEN
   assertS2 KINET :: 'CTRL-FLUX' 1 3.971426E+02 ;
   assertS2 KINET :: 'CTRL-PREC' 1 1.200883E-01 ;
   assertS2 KINET :: 'E-POW' 1 2.047011E+04 ;
 ELSEIF TIME 26.7 - ABS 1.0E-3 < THEN
   assertS2 KINET :: 'CTRL-FLUX' 1 4.351272E+02 ;
   assertS2 KINET :: 'CTRL-PREC' 1 1.224600E-01 ;
   assertS2 KINET :: 'E-POW' 1 2.245449E+04 ;
 ENDIF ;
 MACRO1 SYSTEM1 := DELETE: MACRO1 SYSTEM1 ;
 MACRO1 := MACRO2 ;
 SYSTEM1 := SYSTEM2 ;
 MACRO2 SYSTEM2 := DELETE: MACRO2 SYSTEM2 ;
ENDWHILE ;
ECHO "test lmw2D completed" ;
```

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 $i_{src1}, 32, 33$ $i_{src2}, \, 32$ lreg, 10::, 2, 13, 18, 21, 24, 25, 27, 30, 32, 34, 36, 38,41, 44 :=, 2, 13, 18, 21, 24, 25, 27, 30, 32, 34, 38, 41, 44 ;, 1 ACCE, 27, 28, 32, 33, 42 ACYL, 4, 5 $\mathtt{ADI},\, 21,\, 23,\, 27,\, 28,\, 32,\, 33,\, 42,\, 43$ ADJ, 27, 28, 30, 31 ADJO, 34, 35 ALBE, 4, 5albedo, 4, 5albedp, 13, 14 ALBP, 13, 14ALL, 9, 32, 33 ANG, 4, 8ang, 4, 8 ANIS, 13, 14 BETA, 38, 39 beta, 38, 39 (BIVACA:), 24 BIVACA:, 24 (bivaca_data), 24 (BIVACT:), 18 BIVACT:, 18 (bivact_data), 18 CAR1D, 3, 11 $\texttt{CAR2D},\, 3,\, 5,\, 8,\, 11$ CAR3D, 3, 5, 8, 10, 11 CHDL, 15, 16 CHI, 15, 16 CHID, 38, 39 chid, 38, 39 COMPLETE, 4, 5, 9CRANK, 42CROWN, 9 CYLI, 4, 5DELP, 13, 14 DELTA, 42delta, 42(DELTA:), 30 DELTA:, 30(delta_data), 30 DERI, 25(descBC), 3, 4(descMC), 3, 4, 9(descPOS), 3, 4, 10(descval), 44

DIAG, 4, 5, 8, 10 DIFF, 15, 16, 18, 19, 21, 22 diff, 15, 16 DIFFX, 15, 16 DIFFY, 15, 16 DIFFZ, 15, 16 DIM, 44 dim, 44 DIRE, 34DMACRO, 25, 26 DOLD, 13, 15 DSYST, 30 DUAL, 18, 19, 21, 22 dxyz, 44, 45 EDIT, 3, 4, 13, 18, 21, 24, 25, 27, 30-34, 38, 39, 42, 44 END: ;, 1 epsout, 27, 28, 32, 33, 42, 43 epsthr, 27, 28, 32, 33, 42, 43 (ERROR:), <u>36</u> ERROR:, 36EXPLICIT, 32, 33 EXPON, 42EXTE, 27, 28, 32, 33, 42, 43 FIXE, 15, 16 (FLUD:), 27 FLUD:, 27 (flud_data), 27 FLUNAM, 44 FLUX, 27-29, 34, 38, 39 FLUX, 42FLUX0, 30, 32 FLUX_GPT, 32, 33 FLUXUNK, 44fnorm, 39 FROM-TO, 32, 33 (GEO:), 2 GEO:, 2 GEO: ::, 2 (geo_data1), 2, 3 (geo_data2), 2, 3 GEOM, 18, 21, 34 GEOM1, 2GEOM2, 2GPT, 30, 32, 33 (GPTFLU:), 32 GPTFLU:, 32 (gptflu_data), 32 H-FACTOR, 15, 16 HBC, 4

HEX, 3, 11 HEXZ, 3, 11 hfact, 16 hname, 36, 37 HOMOGE, 3 HREA, 36, 37

i, 9 ic, 9icl1, 27, 28, 32, 33, 42 icl2, 27, 28, 32, 33, 42 *icode*, **4**, **5** icol, 18-23 ielem, 18-23 *IFLU*, **44** ihom, 34, 35 ilastg, 15, 17 imix, 9 imode, 34, 44 IMPLIC, 42IMPLICIT, 32, 33 impv, 21, 23 IN, 34, 35 (INIKIN:), 38 INIKIN:, 24, 26, 38 (inikin_data), 38, 39 INPUT, 13INTERPFLUX, 44INTG, 34, 35 iplan, 9 iplan1, 9 iprint, 3, 4, 13, 18, 21, 22, 24, 25, 27, 30-34, 38, 39, 42, 44 iscat, 18, 19, 21, 22 iseg, 21, 23 isplh, 18, 19, 21-23 ispltr, 10, 11 ispltx, 10, 11 isplty, 10, 11 ispltz, 10, 11 istep, 13, 14 jmix, 9 KINET, 38, 41 (KINSOL:), 41 KINSOL:, 24, 26, 41 (kinsol_data), 41, 42 LAMBDA, 38, 39 lambda, 38, 39 lc, 9lh, 3, 4, 11 LINKED_LIST, 1 lmod, 27, 28 *lp*, 9

lr, 3, 4, 10 LUMP, 21, 22 *l*x, **3**, 10 ly, 3, 4, 10 lz, 3, 4, 10 (MAC:), 13 MAC:, 5, 13 $(mac_data), 13$ MACLIB, 14 MACR1, 13, 14 MACR2, 13 MACRO, 24-27, 34, 38, 41 MACRO1, 36, 37 MACRO2, 34, 36, 37 MACRO_0, 41 (macxs), 13-15matnum, 15, 16 MAX, 39maxout, 27, 28, 32, 33, 42, 43 maxpts, 11, 18, 19, 21, 22 MAXR, 18, 19, 21, 22 maxthr, 27, 28, 32, 33, 42, 43 MCFD, 18, 19, 21, 22 MESHX, 10MESHY, 10 MESHZ, 10MIX, 9, 15, 34, 35 MODE, 34, 44 MODULE, 1 MONI, 27, 28 n, 18, 19, 21-23 nadi, 21, 23, 27, 28, 32, 33, 42, 43 nalbp, 13, 14 NAME1, 1 NAME2, 1 NAME3, 1 NAME4, 1 NAME5, 1 naniso, 13-15, 17 nbscat, 15, 17 NDEL, 38, 39 ndel, 13–15 ndg, 13, 14, 38, 39 NGRO, 13, 14 ngroup, 13-15, 17 NGRP, 38, 39 ngrp, 38, 39 NIFI, 13, 14 nifiss, 13-15NMIX, 13, 14 nmixt, 9, 13–15 NORM, 39 nrads, 4, 8NREG, 36, 37

nreg, 36, 37 NTOTO, 15 NTOT1, 15 NUSIGD, 15, 16 NUSIGF, 15, 16 nvd, 18, 19, 21, 23 OLD, 13, 14 (OUT:), 34 OUT:, 34 $(out_data), 34$ OVEL, 25, 26 OVERV, 15, 16 overv, 15, 16 PERT, 25, 26 PLANE, 9 PN, 18, 19 power, 34, 35, 39 POWER-INI, 39 POWR, 34, 35 PREC, 42PRIM, 18, 19, 21, 22 PRTV, 21, 23 R+, 4 R120, 4, 5R180, 4, 5RADIUS, 10 RADS, 4, 8 RAND, 27, 28 READ, 13, 14 REFL, 4, 5RELAX, 27, 29 relax, 27, 29 rrad, 4, 8 *rrr*, **10** S30, 4, 5 S90, 4, 5SA180, 4, 5 SA60, 4, 5 SAME, 9SB180, 4, 5 SB60, 4, 5 SCAT, 15, 16, 18, 19, 21, 22 scat, 15SCHEME, 42SEQ_ASCII, 1 SEQ_BINARY, 1 SIDE, 10 sidhex, 10 SKIP, 25(specif), 1SPHERE, 3, 10, 11 SPLITR, 10

SPLITX, 10 SPLITY, 10 SPLITZ, 10 SPN, 18, 19, 21, 22 STEP, 13, 14 SYME, 4, 5 SYST, 24-27, 32, 38, 41 SYST0, 30 SYST_0, 41 THER, 27, 28, 32, 33, 42, 43 THETA, 42theta, 42TITL, 18, 19, 21, 22 TITLE, 18, 19, 21, 22 TOTAL, 15TRACK, 18, 21, 24, 25, 27, 28, 30, 32, 34, 38, 41, 43 TRACKING, 44 TRAN, 4, 5, 8TRANC, 15(trip2), 13, 15 (triv2), 13, 14 (TRIVAA:), 25 TRIVAA:, 25(trivaa_data), 25 (TRIVAC), 1 (TRIVAT:), 21 TRIVAT:, 21 (trivat_data), 21 TRKNAM, 44 ttflx, 42ttprc, 42 TUBE, 3, 10, 11TUBEZ, 3, 10, 11 UNIT, 24-26 UPTO, 9(VAL:), 44 VAL:, 44 VAR1, 27, 28, 32, 33, 42 VECT, 21, 23 VOID, 4, 9, 18, 19, 21, 23 X+, 4, 8–10 X-, 4, 8-10 xdiffx, 15, 16 xdiffy, 15, 16 xdiffz, 15, 16 xhfact, 15 xrad, 4, 8 xschi, 15, 16 xschid, 15, 16 xsfixe, 15, 16 XSM_FILE, 1

xsscat, 17 xssig1, 15 xssig1, 15, 16 xssigt, 15, 16 xssigt, 15 xsstra, 15, 16 xxx, 10 Y+, 4, 8–10 Y-, 4, 8–10 yyy, 10 Z+, 4, 8–10 Z+, 4, 8–10 Z=, 4, 8–10 ZERO, 4, 5 zzz, 10